



SMILE project and the current status

A. Takada (Kyoto Univ.)

1. MeV gamma ray Imaging & ETCC
2. Results of SMILE-I
3. Preparations of SMILE-II

Motivation

Observation of MeV gamma-ray will provide us...

◆ Nucleosynthesis

SNR : Radio-isotopes

Galactic plane : ^{26}Al • ^{60}Fe
Annihilation

◆ Acceleration

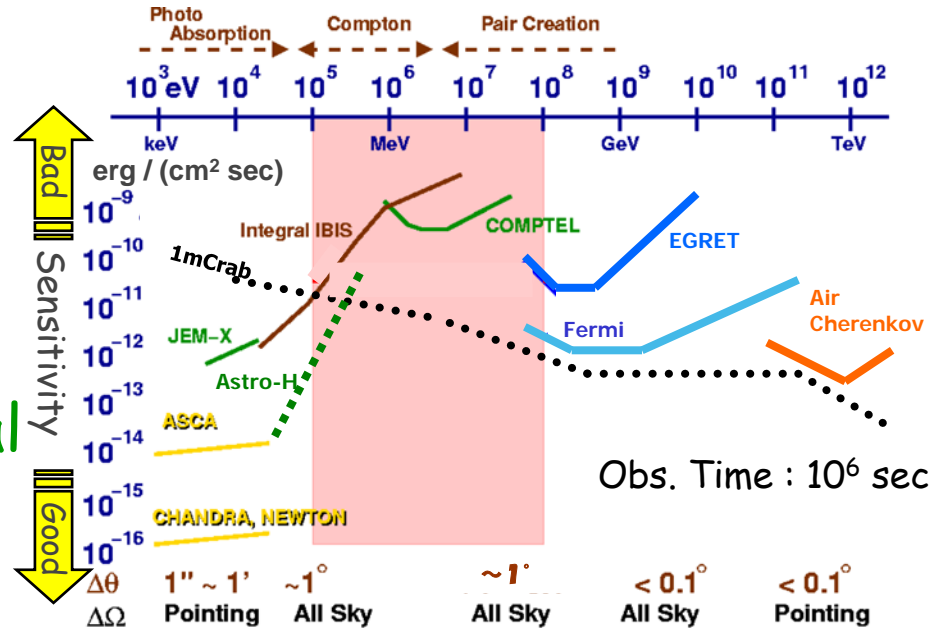
Jet (AGN) : Synchrotron
+ Inverse Compton

◆ Strong Gravitational Potential

Black Hole : accretion disk, π^0

◆ Etc.

Gamma-ray Pulsar, solar flare



- The observation of continuum component is also important.
- Where are MeV gamma-ray objects?
- There are many background events which obstruct the observations.

Requirements for the next-generation detectors are ...

- Wide-band detection
- Large Field of View
- Background rejection

Sky Map of MeV Gamma rays

COMPTEL(1-30MeV) 32 objects

AGN 10

Line Emissions from SNR 7

Crab 1 γ -Pulsar 3

B.H.Candidates 2,

UnID 9

Integral Point Sources

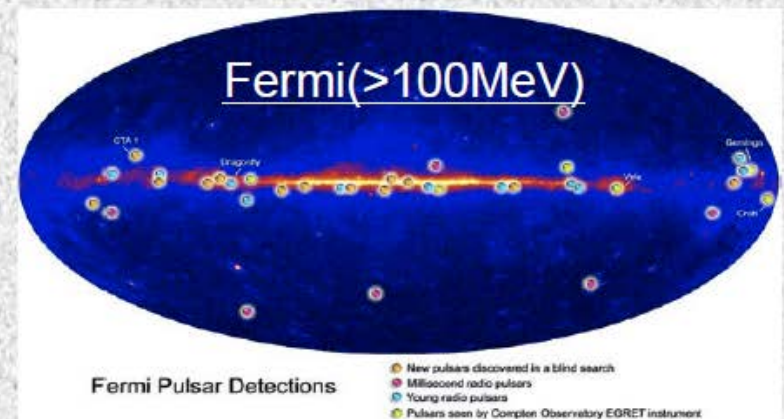
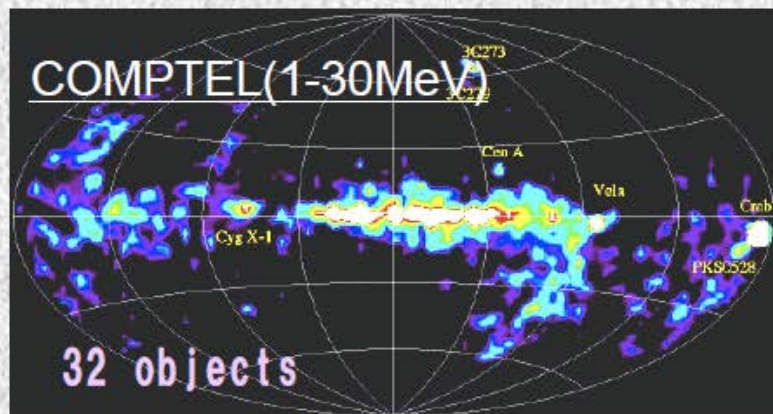
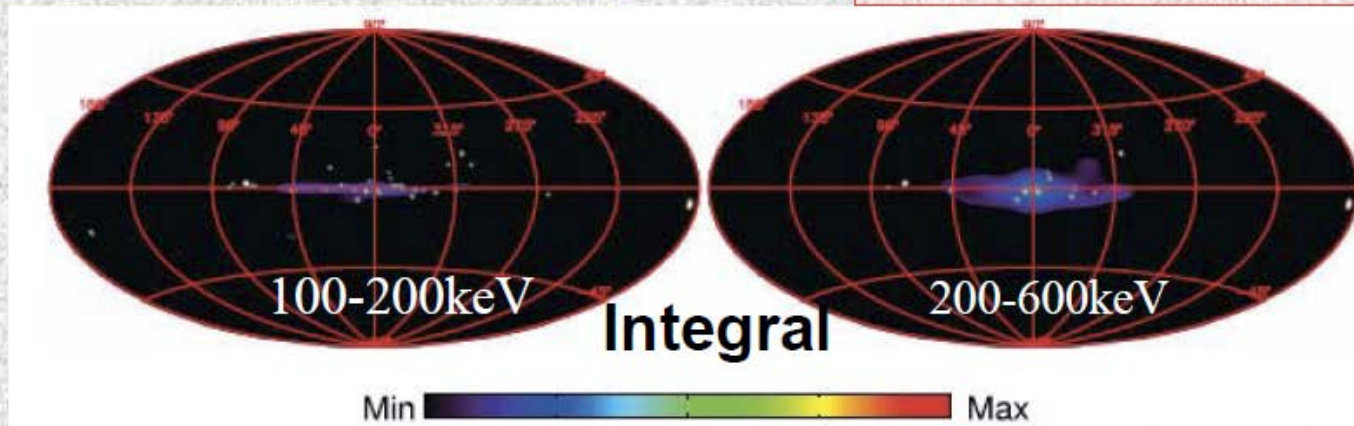
25-50keV 173

50-100keV 79 ($>3.5\sigma$)

100-200keV 30($>2.5\sigma$)

200-600keV 12

>600 keV 4



MeV gamma-ray imaging

➤ Collimator + position-sensitive detector

⇒ OSSE (*CGRO*)

- narrow FOV
- background from collimator
- Energy < 1MeV

➤ Coded Aperture Imaging

⇒ BAT (*SWIFT*), SPI (*INTEGRAL*)

- wide FOV & good angular resolution
- photons from outside of mask are background
- imaging of the spread sources is difficult
- Energy < 1MeV

➤ Laue Lens

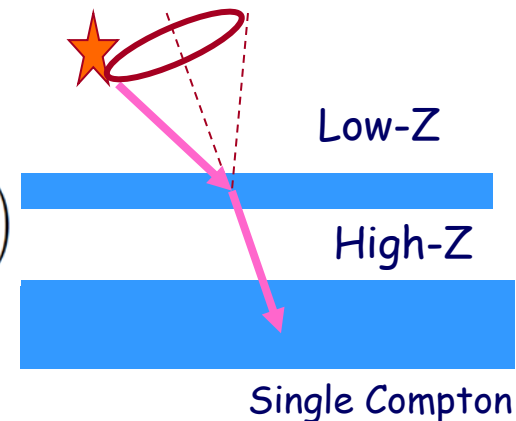
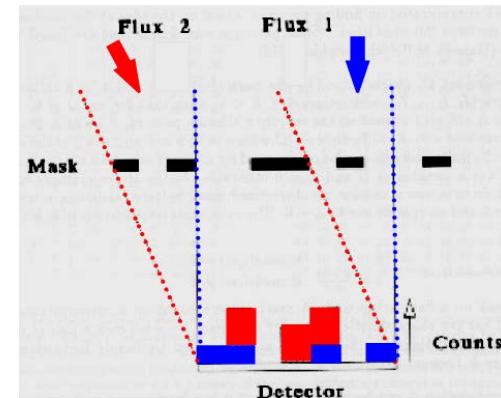
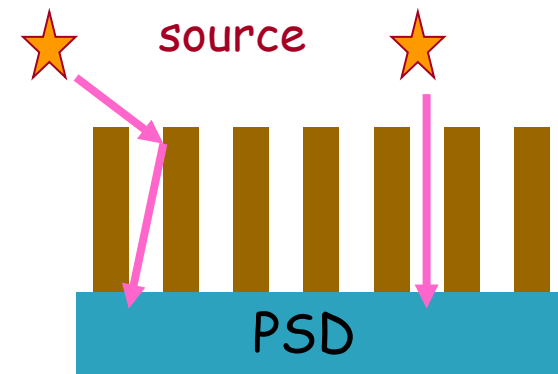
- narrow FOV & good angular resolution
- mono-energy

➤ Compton imaging

⇒ COMPTEL (*CGRO*)

- wide Energy band
- only event circle
- no background rejection

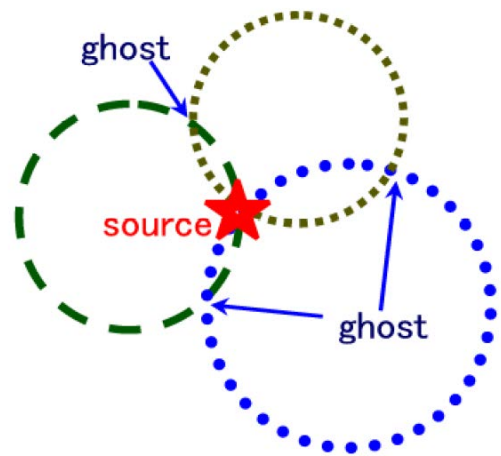
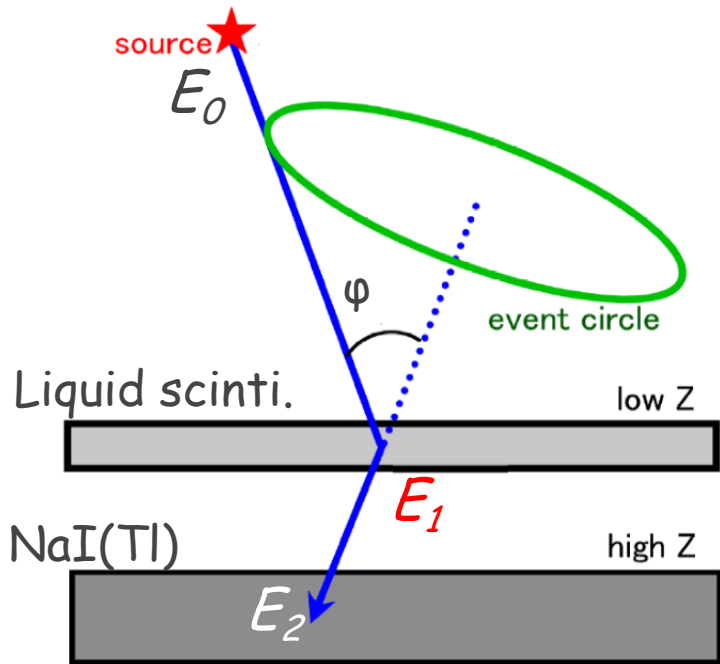
$$\cos \phi = 1 - m_e c^2 \left(\frac{1}{E_2} - \frac{1}{E_1 + E_2} \right)$$



A new method with satisfying requirements

COMPTEL (CGRO:1991~2000)

Using Compton Scattering



- ◆ energies of scattered gamma and recoil electron

➔ Energy of incident gamma
Scattering angle

$$\cos \phi = 1 - m_e c^2 \left(\frac{1}{E_2} - \frac{1}{E_1 + E_2} \right)$$

- ◆ Compton scattering point & Absorption point

➔ Direction of scattered gamma

- ◆ ignore the direction of recoil electron

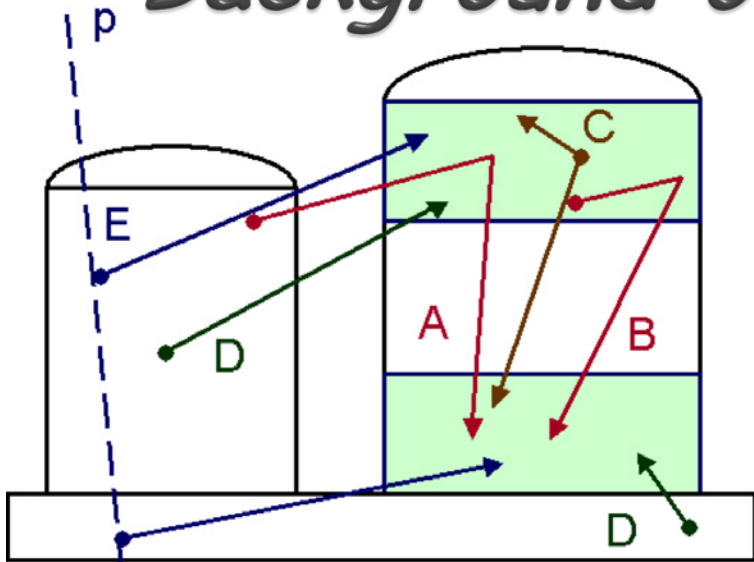


- Restrict the direction of incident gamma-ray to a circle
- The source position is determined fully by piling up circles

➔ require 3 γ at least

Background of COMPTTEL

G. Weidenspointner, et.al. (A&A, 2001)



A : external γ
B : internal γ } Intrinsic background

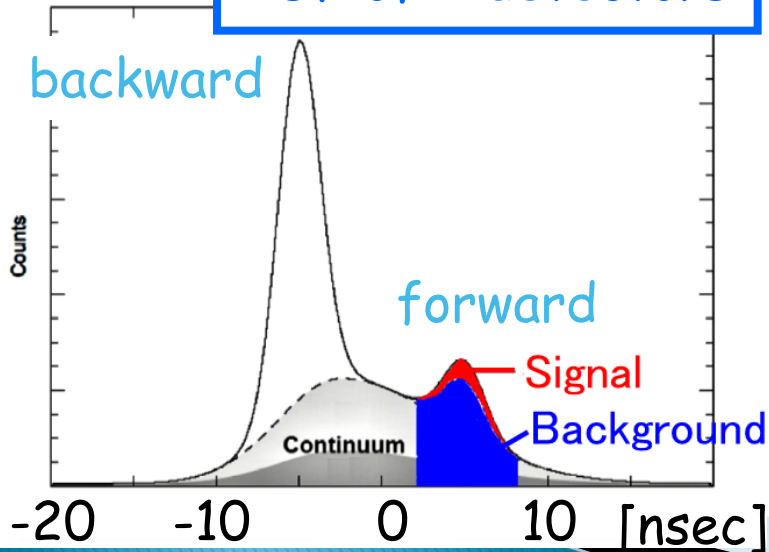
C : two γ

D : random coincidence

E : proton-induced γ

Other background
neutron
electron
gamma from atmosphere

TOF of 2 detectors



COMPTTEL has rejected such background by the measurement of the Time Of Flight between 2 detectors.



Background rejection was not complete
Bad S/N

Electron-Tracking Compton Camera (ETCC)

MeV γ -ray

Drift plane

e^-

μ -PIC

incident γ

Scintillator

PMTs

recoil e

α

scattered γ

- **Gaseous TPC : Tracker**
track and energy
of recoil electron
- **Scintillator : Absorber**
position and energy
of scattered gamma



Reconstruct Compton scattering
event by event

- ▶ 1 photon \Rightarrow direction + energy
- ▶ Large FOV ($\sim 3\text{str}$)
- ▶ **Kinematical background rejection**

$$\cos \alpha_{\text{geo}} = \vec{g} \cdot \vec{e} \iff \cos \alpha_{\text{kin}} = \left(1 - \frac{m_e c^2}{E_\gamma}\right) \sqrt{\frac{K_e}{K_e + 2m_e c^2}}$$

E_γ : Energy of scattered gamma-ray

K_e : Kinematic energy of recoil electron

$m_e c^2$: Rest mass of electron

g : unit vector of scattering direction

e : unit vector of recoil direction

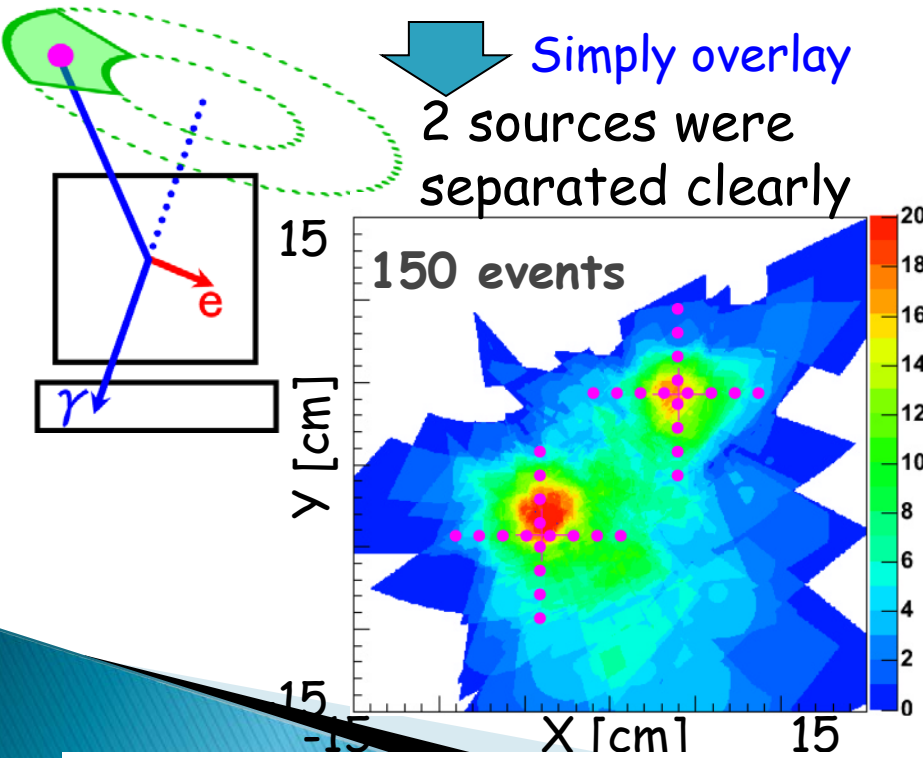
Comparison with the usual Compton method

Electron-Tracking Compton (ETCC)

Using the electron tracks

- complete direction within **sector form** error region

Simply overlay
2 sources were
separated clearly



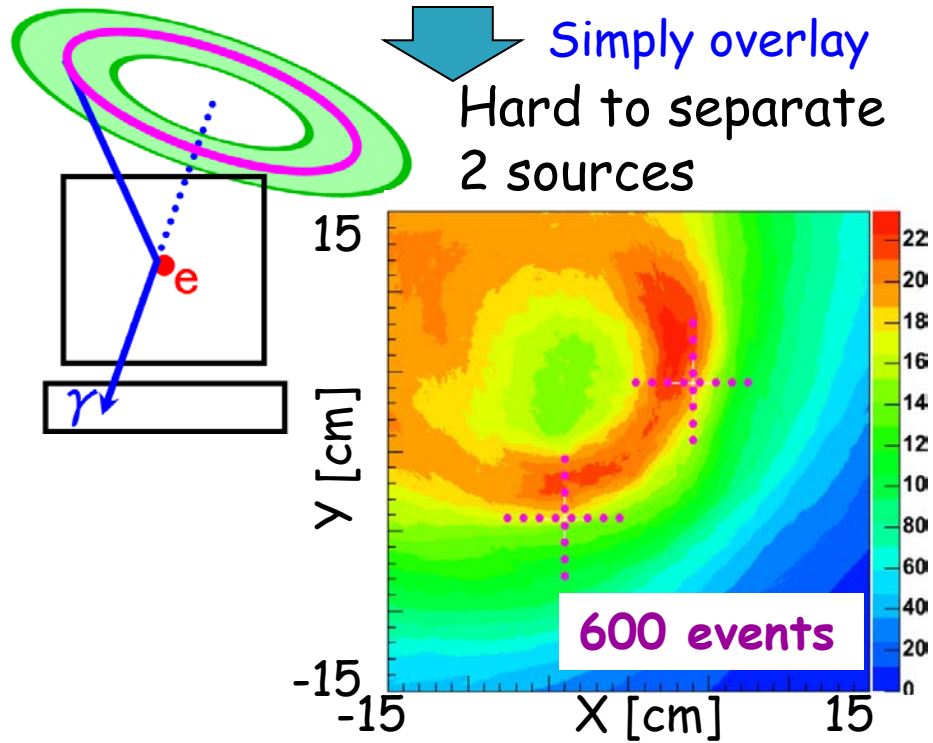
$^{137}\text{Cs}(1\text{MBq})\times 2$, Advanced Compton

Usual Compton Imaging (COMPTTEL)

Not using the electron tracks

- only event circle within **ring form** error region

Simply overlay
Hard to separate
2 sources

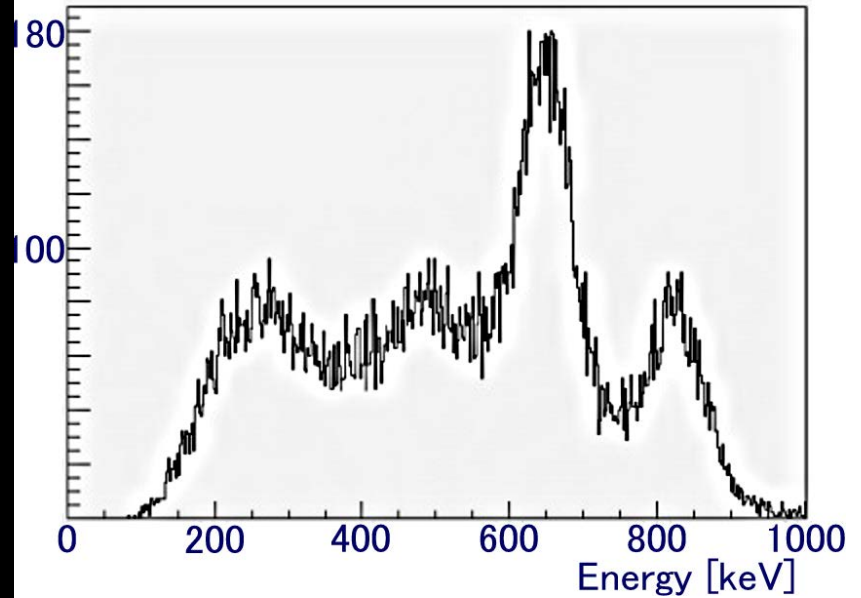
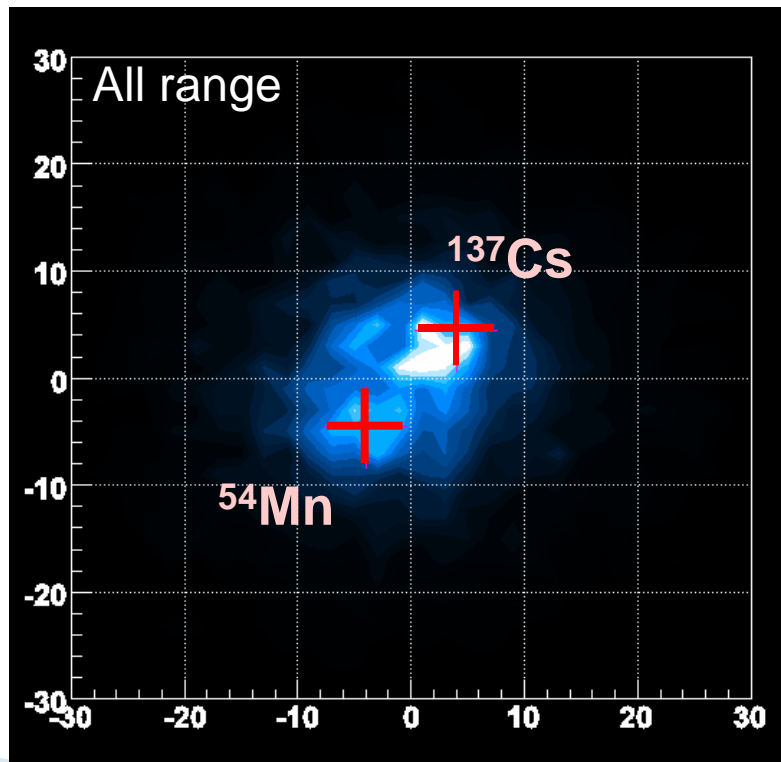


$^{137}\text{Cs}(1\text{MBq})\times 2$, usual Compton

MeV- γ imaging

^{137}Cs : 662keV, 0.89MBq

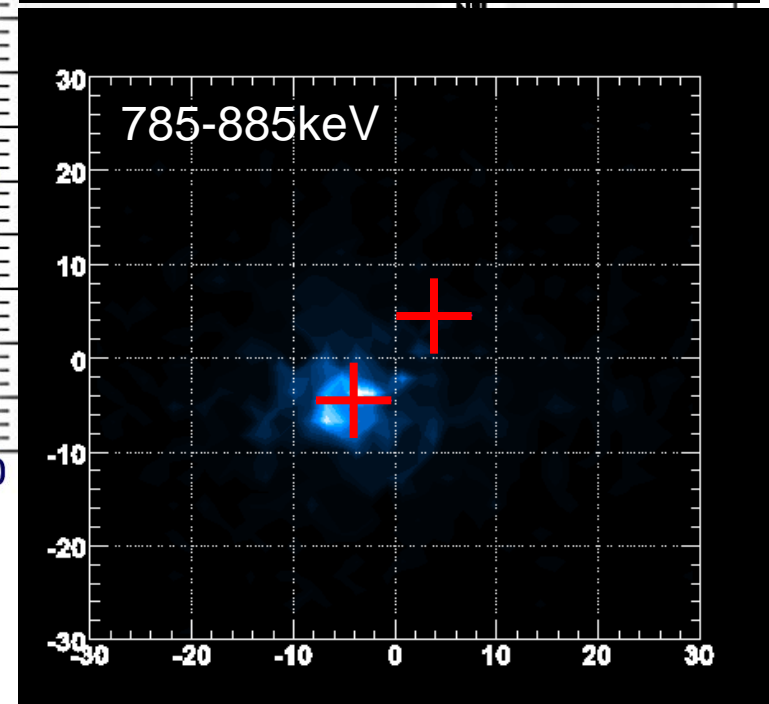
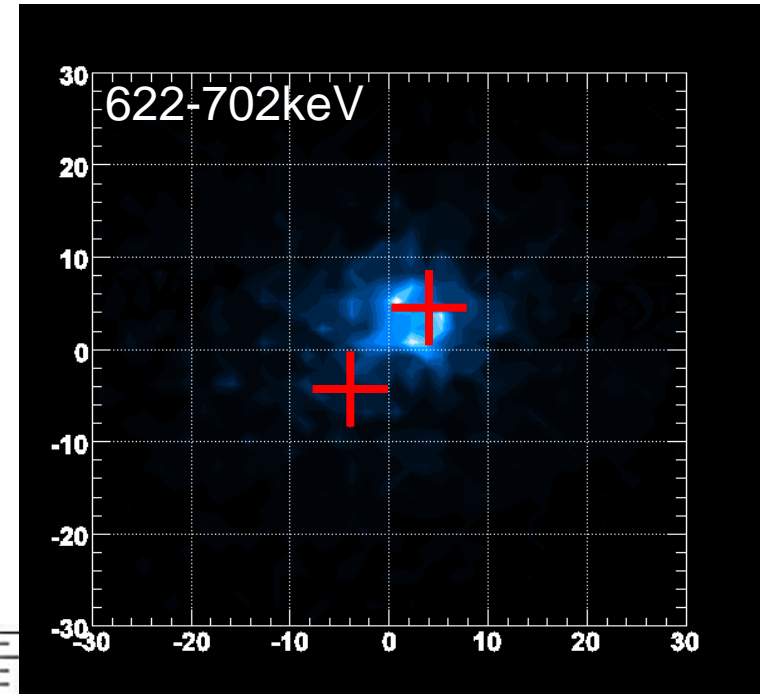
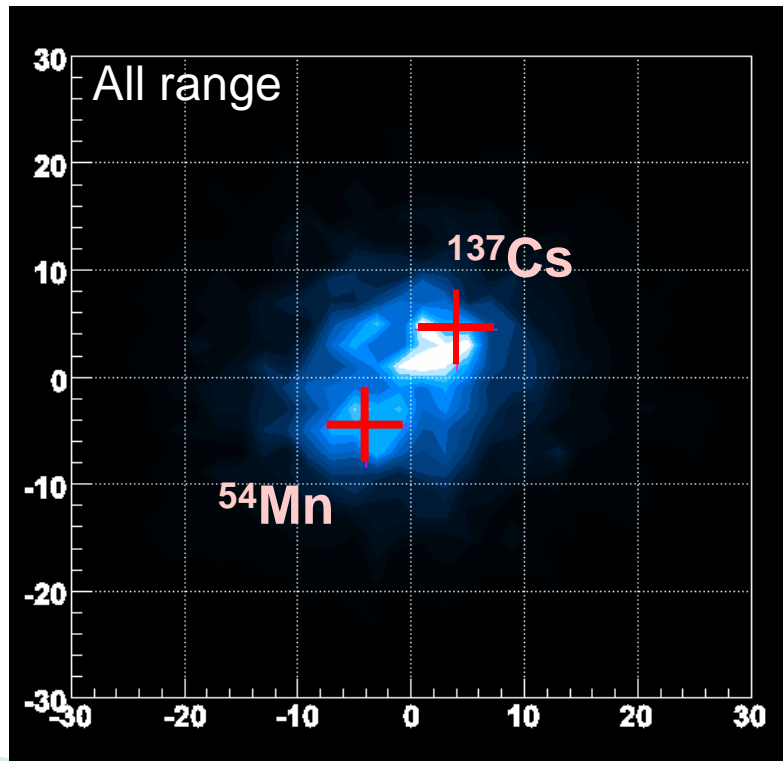
^{54}Mn : 835keV, 0.65MBq



MeV- γ imaging

^{137}Cs : 662keV, 0.89MBq

^{54}Mn : 835keV, 0.65MBq



Sub-MeV gamma-ray Imaging *Loaded-on-balloon Experiment*

10cm cube camera @ Sanriku (Sep. 1st 2006)

- Operation test @ balloon altitude
- Observation of diffuse cosmic/atmospheric gamma
~400 photons during 3 hours
(100 keV~1MeV)

30cm cube camera

- Observation of Crab/Cyg X-1

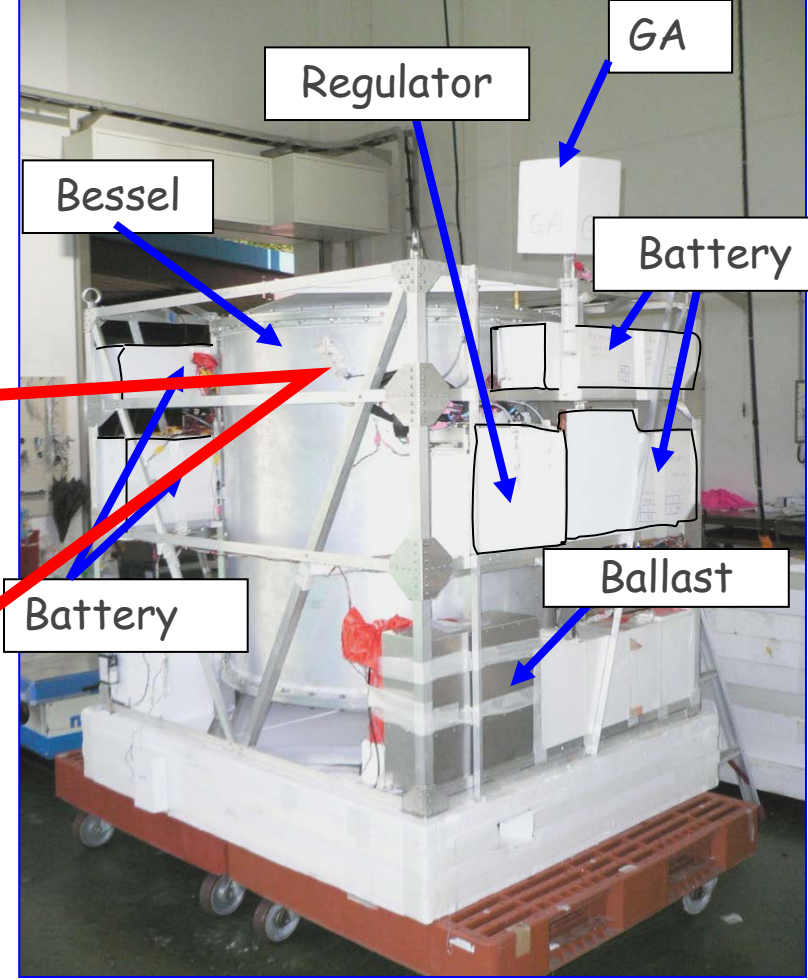
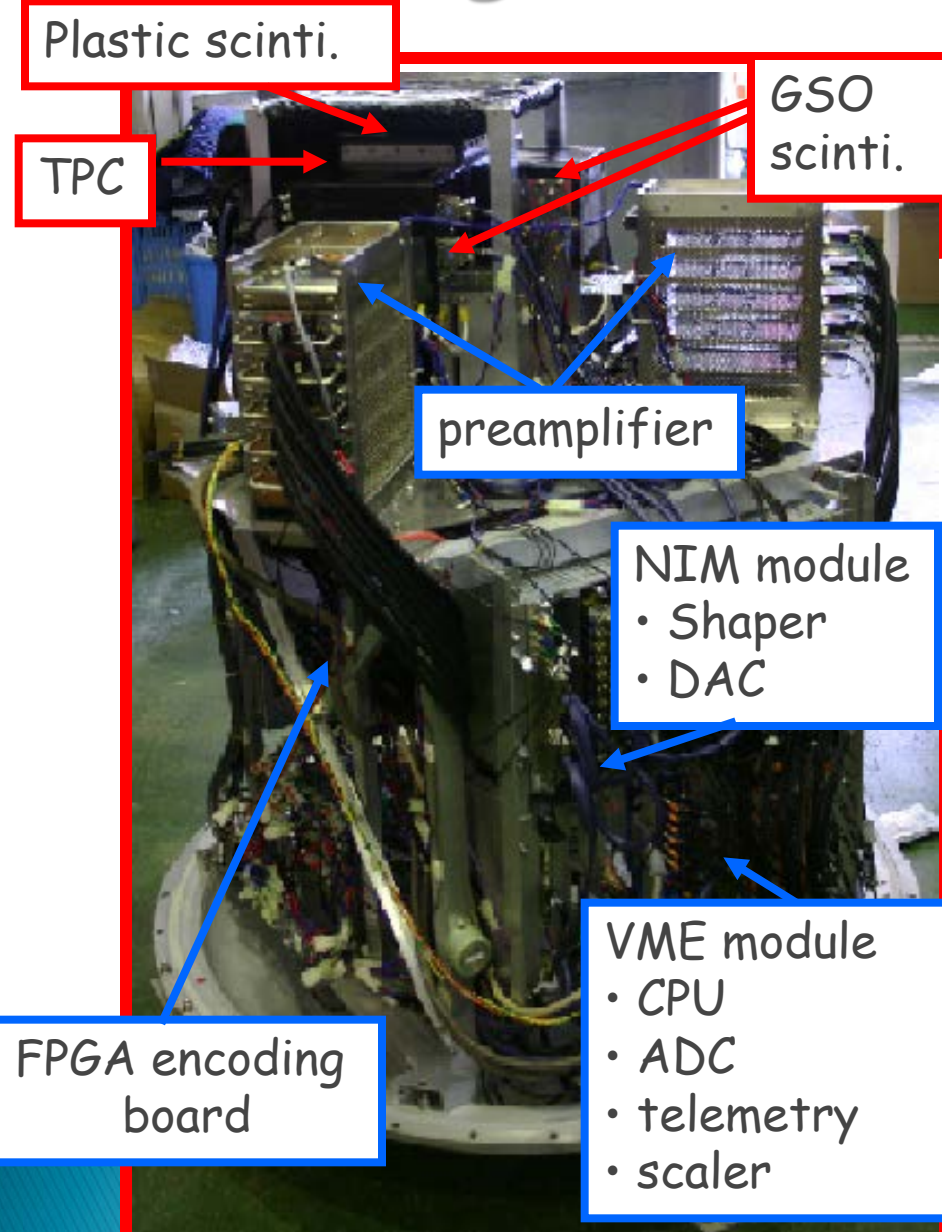
40cm cube camera Sub-MeV ~ MeV

- Long duration observation with super pressure balloon
- Adding pair-creation mode

50cm cube camera

- All sky survey (load on a satellite)

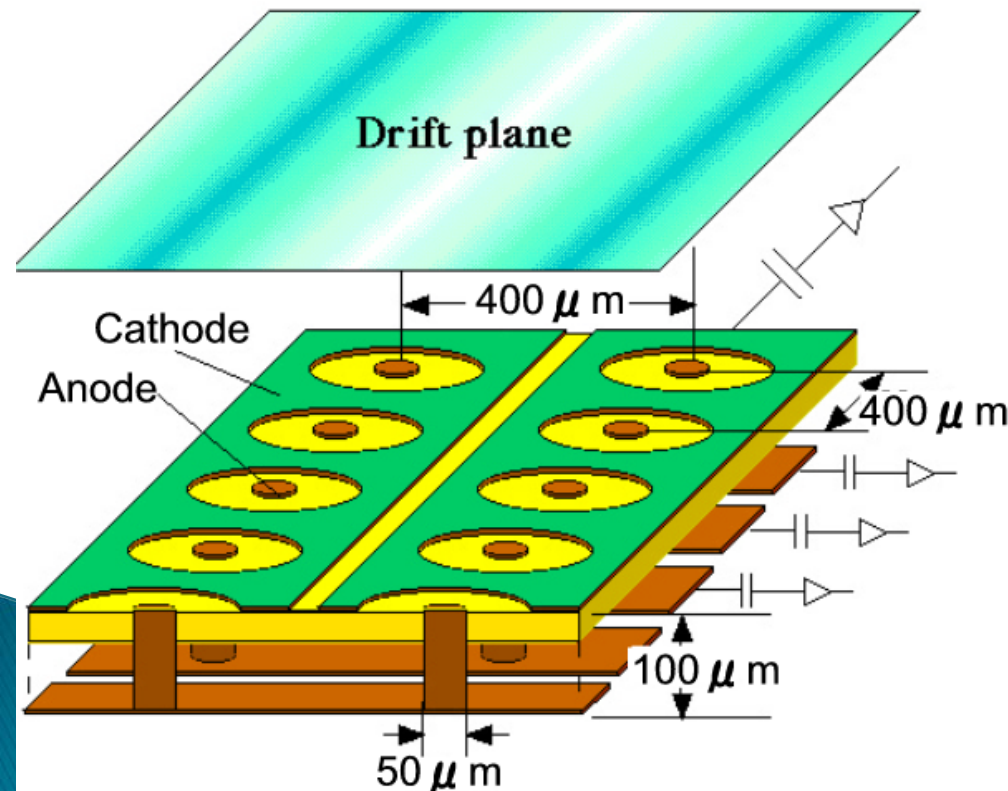
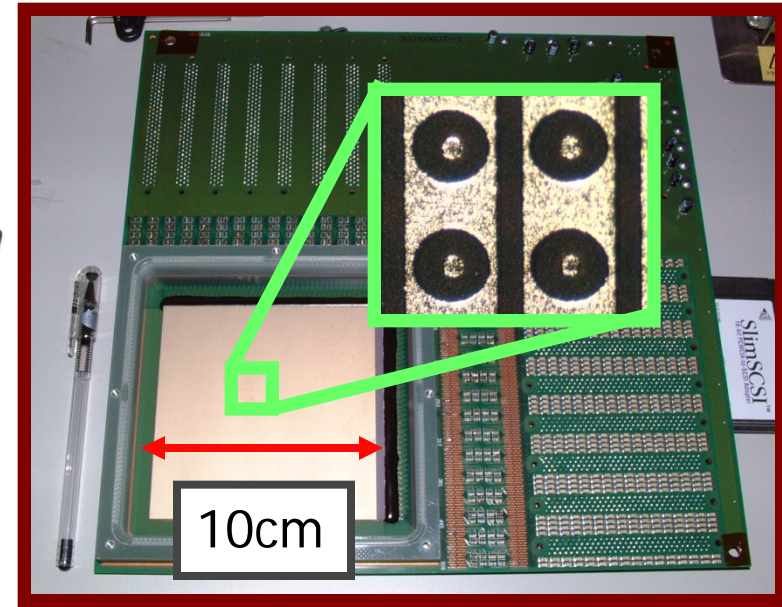
SMILE-I gondola



Size : $1.45 \times 1.2 \times 1.55 \text{m}^3$
Weight : 397 kg
Power : ~250 W
No posture control !!

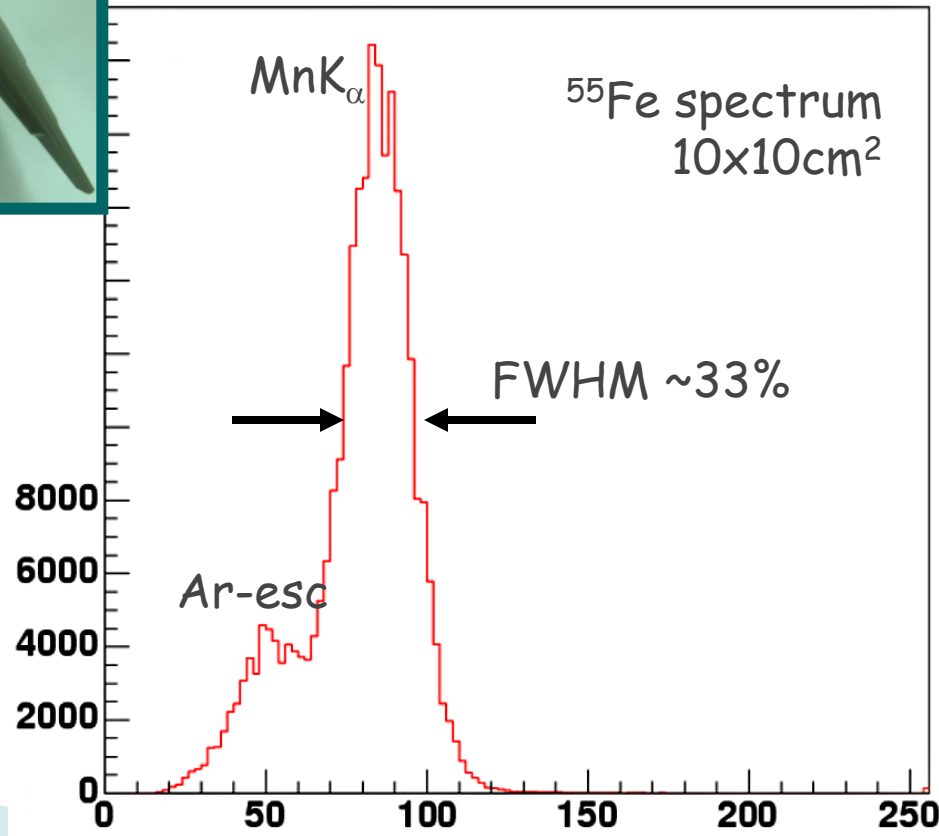
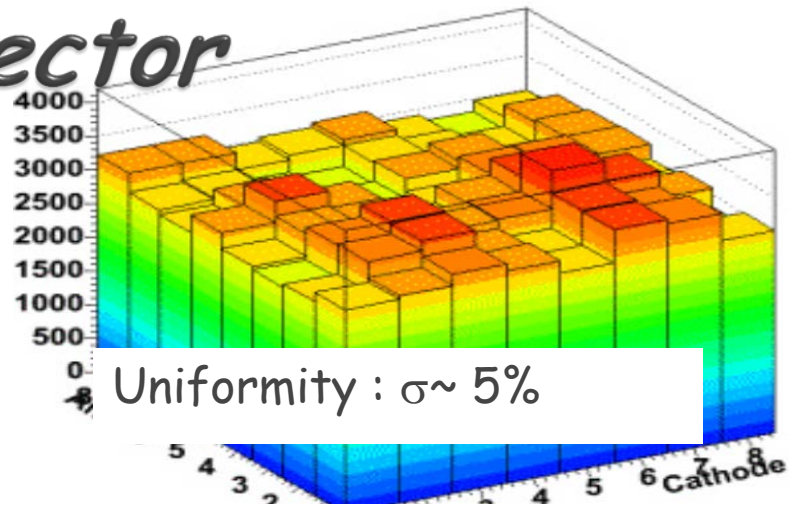
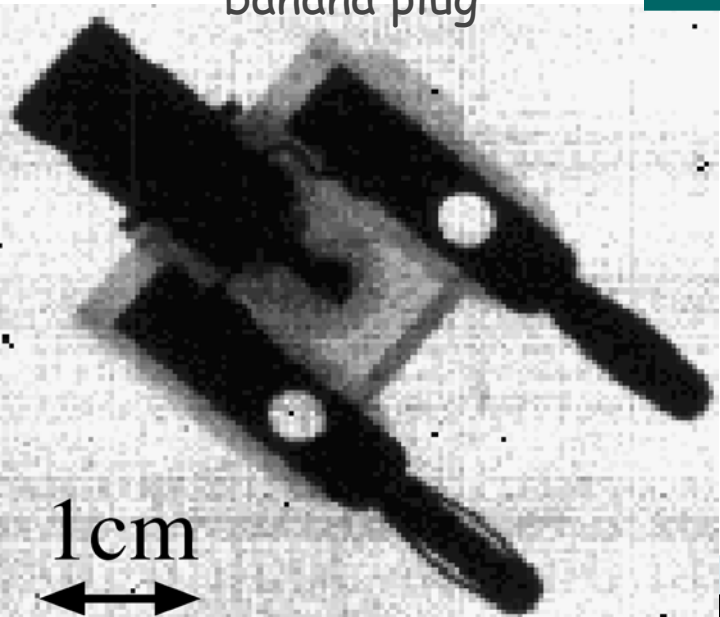
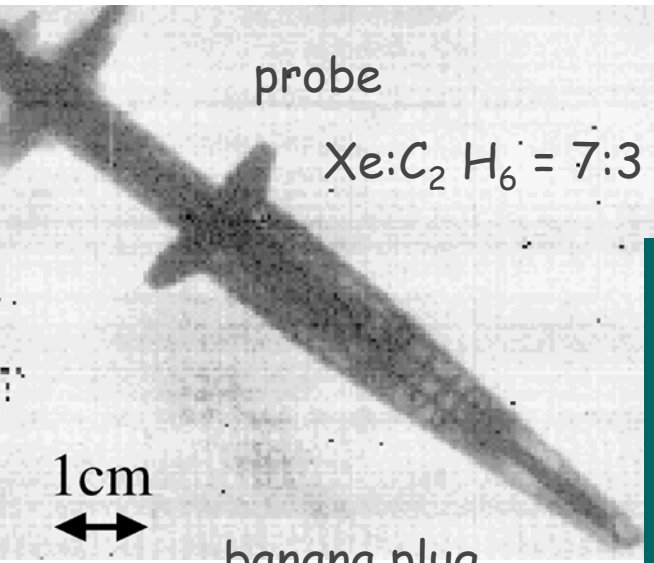
The character and structure of μ -PIC

- 2D readout (~ 65000 pixels)
- Large detection area ($10\text{cm} \times 10\text{cm}$)
- Print Circuit Board technology



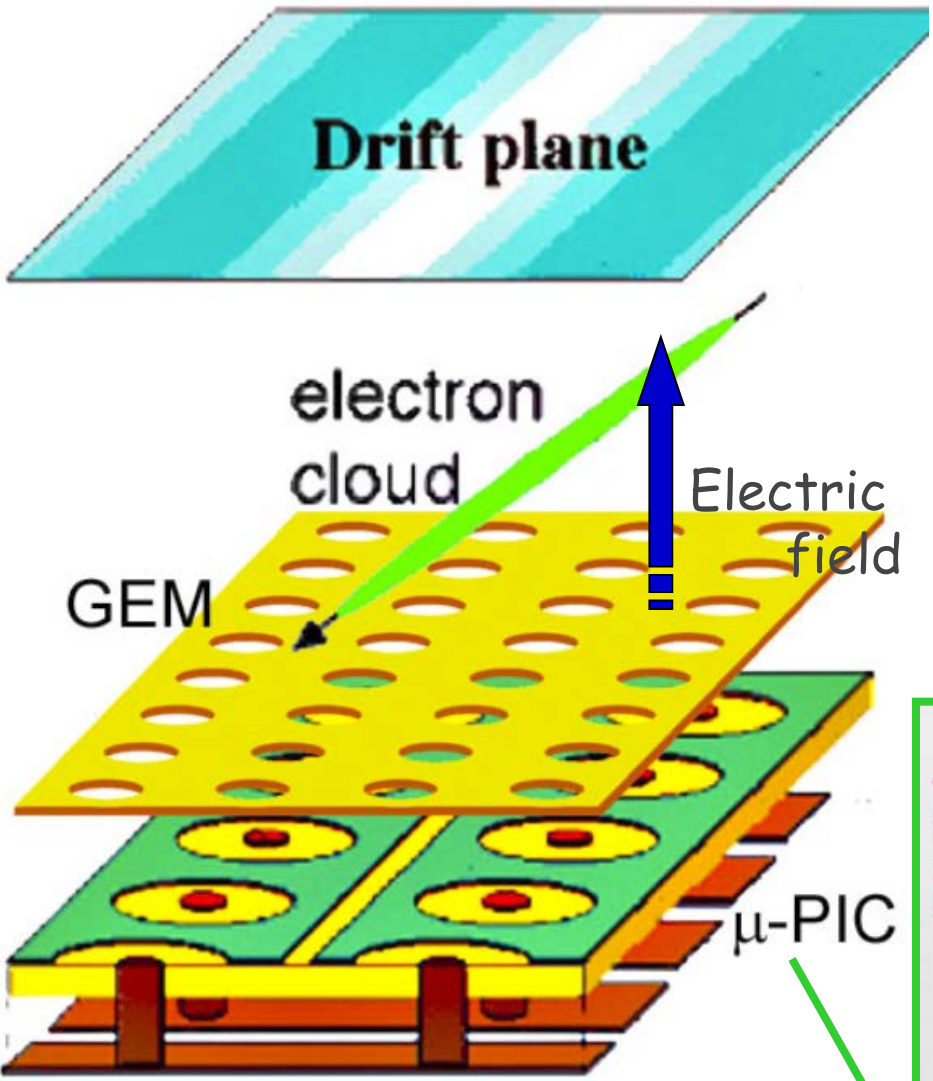
- **max gas gain ~ 16000**
- ▶ energy resolution
30% @ 5.9keV (100cm^2)
- ▶ **stable operation for 1000h**
@ gas gain ~ 6000
- ▶ **good gas gain uniformity**
4.5% @ 100cm^2
- ▶ **fine position resolution**
($\sim 120\mu\text{m}$)

μ -PIC as a X-ray Detector

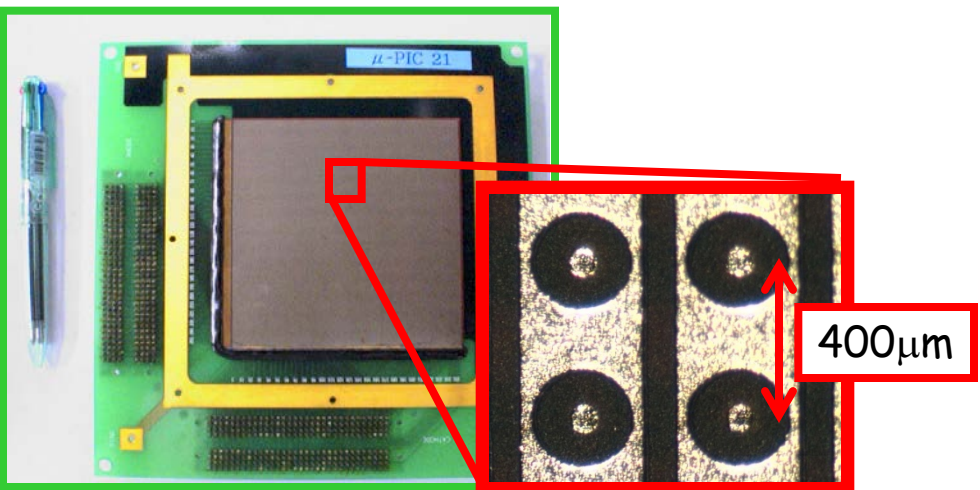


Gaseous electron tracker

2D readout (400 μ m pitch) + Drift time (100MHz)



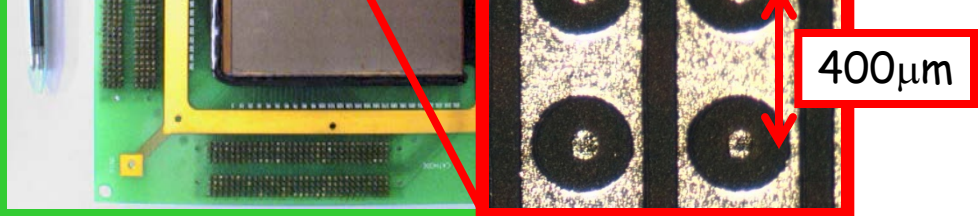
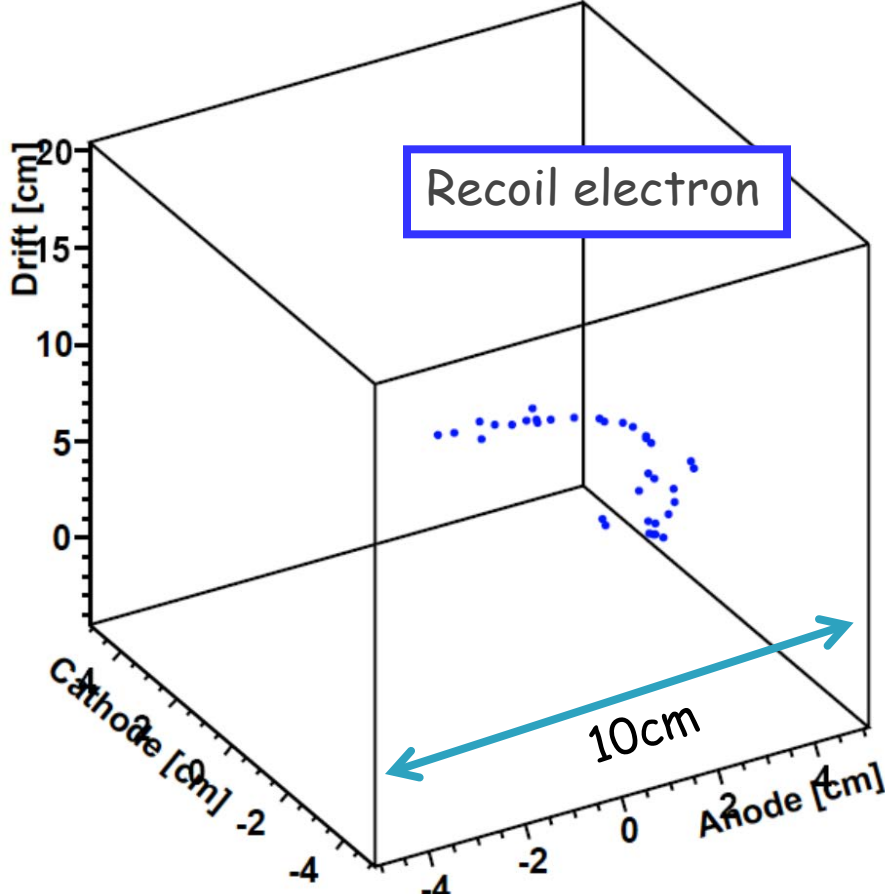
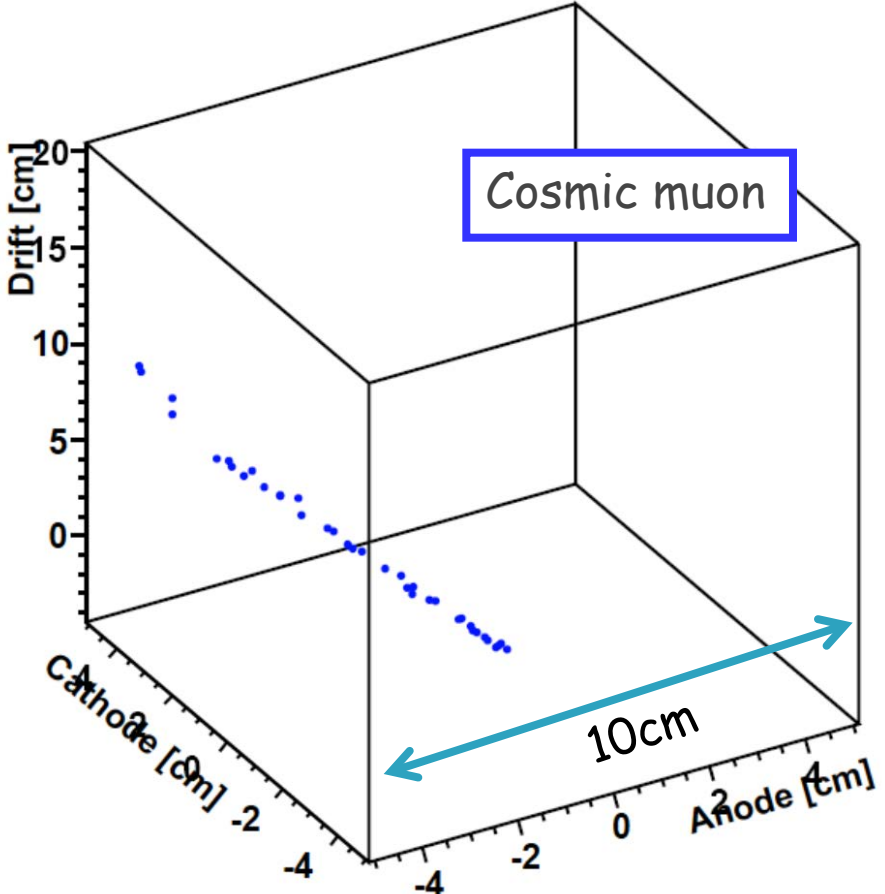
- Gas : **Xe 80% + Ar 18% + C₂H₆ 2%**
1atm, sealed
- Gain : **~35000**
- Drift velocity ($V_d=400V/cm$) :
measured **2.5cm/ μ sec**
simulation **2.48cm/ μ sec**
- Volume : 10 \times 10 \times 14 cm³
- Energy resolution :
~45% (22.2keV, FWHM)
- Position resolution : **~500 μ m**



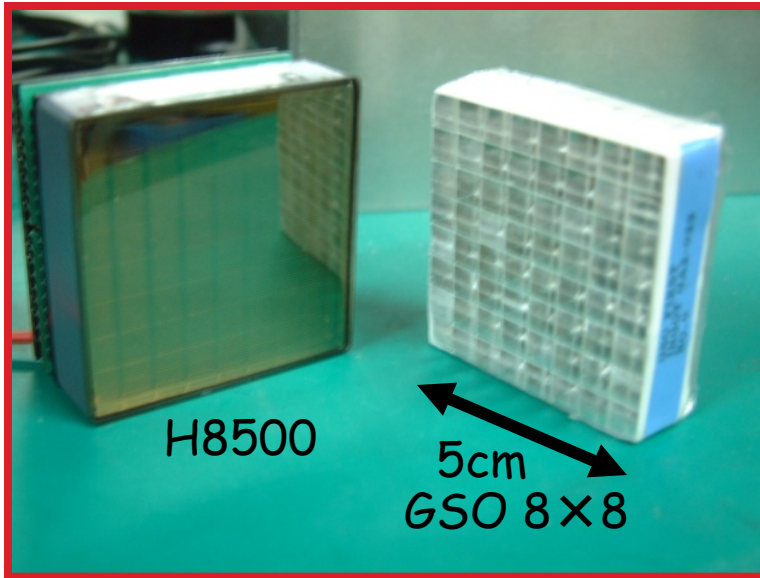
Gaseous electron tracker

2D readout (400 μ m pitch) + Drift time (100MHz)

➤ Gas : Xe 80% + Ar 18% + C₂H₆ 2%

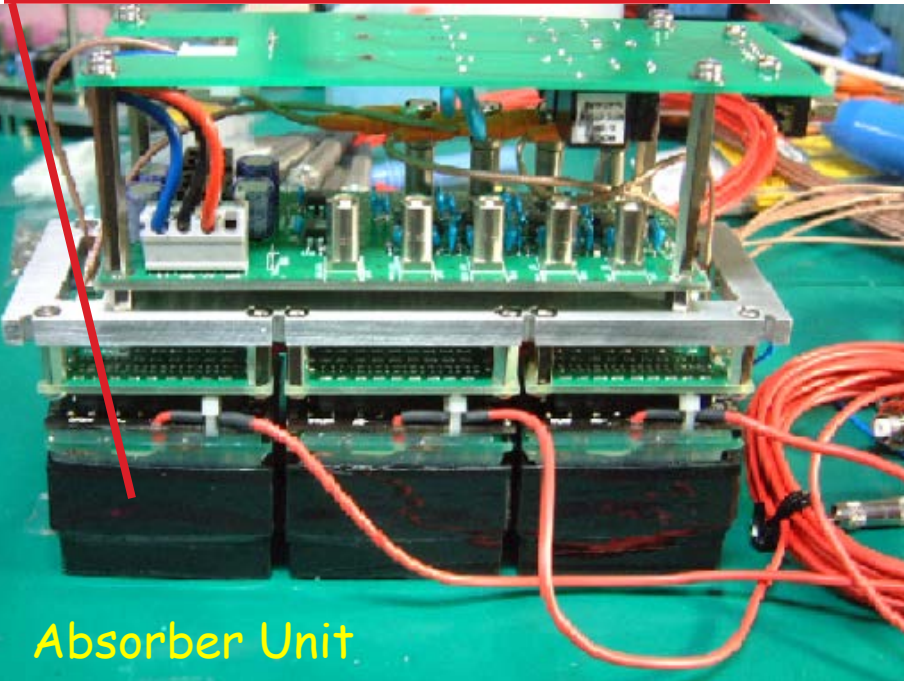
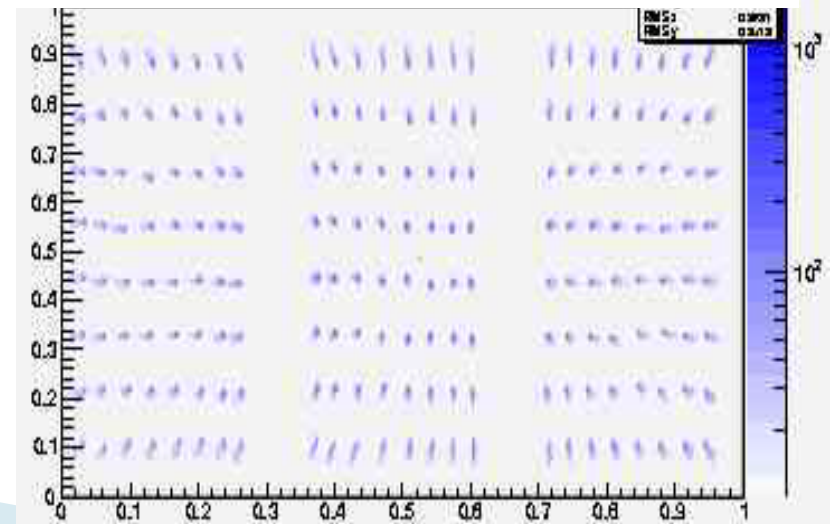


Scintillation Camera

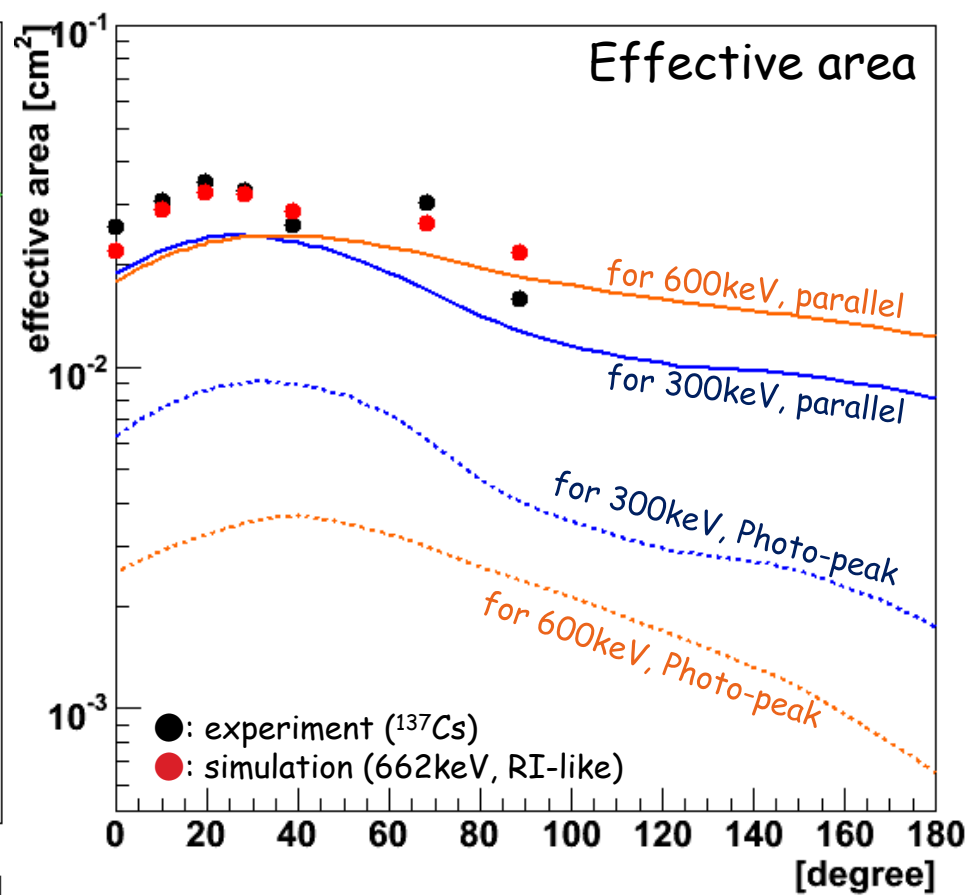
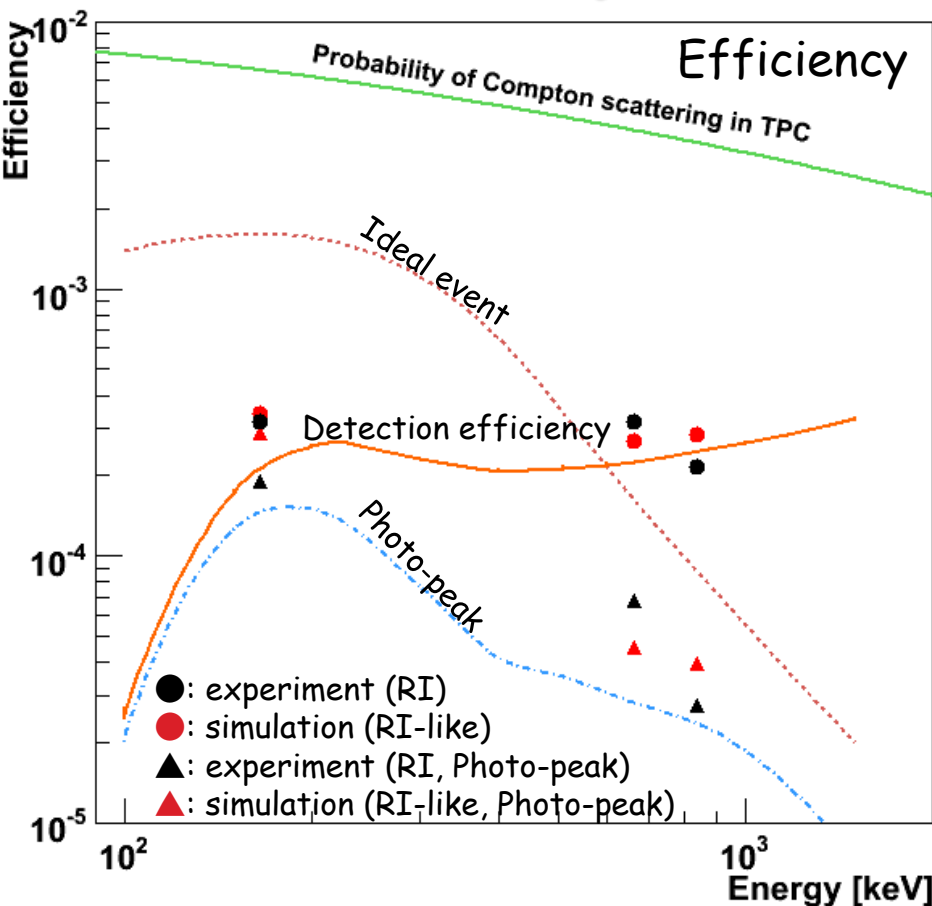


- Scintillator : **GSO(Ce)**
- Pixel size : **6x6x13 mm³**
- Photo readout : H8500 (HPK)
- DC/HV : EMCO Q12N-5
- A unit consists of 192 pixels, 3 PMTs, 3 DC/HV and 4 preamplifier
- 4 channels readout with resistive chain
- Bottom : 3x3 PMTs
- Side : 3x2 PMTs x 4 } **2112 pixels**
- Energy resolution : **~11% (662keV, FWHM)**

¹³⁷Cs Position imaging map



Efficiency & Effective area



- Detection Efficiency : 3×10^{-4} for 150-1500keV
- Effective area : $2 \times 10^{-2} \text{ cm}^2$ for 150-1500keV, 0-60°
- The simulated effective area was roughly consistent with that obtained by experiments.
- Effective area has a maximum at $\sim 25^\circ$ ← caused by the geometry

1st Flight SMILE

- Gondola size: $1.45 \times 1.2 \times 1.55 \text{m}^3$
- Gondola weight: 397kg
- Bessel: $\phi 1 \times 1.4 \text{m}^3$
- Power: $\sim 350 \text{W}$
in Bessel : 220W

In Bessel (1 atm)

Detector, DAQ system,
Storage, Thermometer,
Pressure gauge,
GPS, Clinometer

Out of Bessel

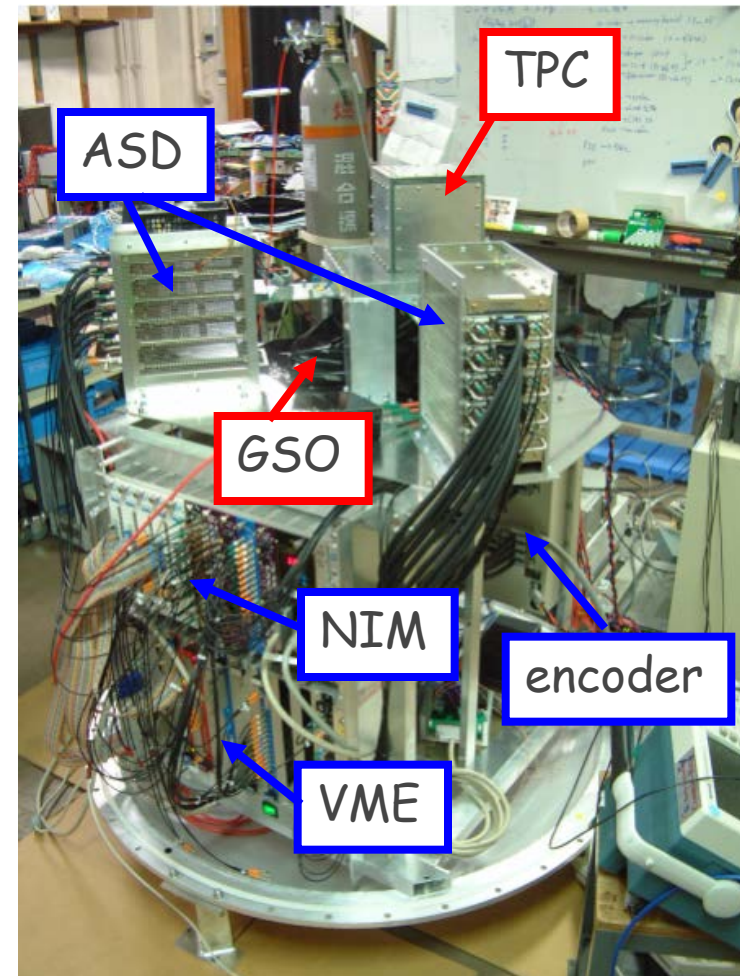
Battery & Regulator,
Thermometer,
Pressure gauge,
GPS antenna,
Geomagnetic aspectmeter

Flight Control

Telemetry,
Transponder,
Buoy, Radiosonde,
GPS, Thermometer,
Pressure gauge, etc.

Balloon

B100 ($100,000 \text{m}^3$)
Weight 816kg
Buoyancy 888.2kg



1st Flight SMILE

- Gondola size: $1.45 \times 1.2 \times 1.55 \text{m}^3$
- Gondola weight: 397kg
- Bessel: $\phi 1 \times 1.4 \text{m}^3$
- Power: $\sim 350 \text{W}$
in Bessel : 220W

In Bessel (1 atm)

Detector, DAQ system,
Storage, Thermometer,
Pressure gauge,
GPS, Clinometer

Out of Bessel

Battery & Regulator,
Thermometer,
Pressure gauge,
GPS antenna,
Geomagnetic aspectmeter

Flight Control

Telemetry,
Transponder,
Buoy, Radiosonde,
GPS, Thermometer,
Pressure gauge, etc.

Balloon

B100 ($100,000 \text{m}^3$)
Weight 816kg
Buoyancy 888.2kg



1st Flight SMILE

- Gondola size: $1.45 \times 1.2 \times 1.55 \text{m}^3$
- Gondola weight: 397kg
- Bessel: $\phi 1 \times 1.4 \text{m}^3$
- Power: $\sim 350 \text{W}$
in Bessel : 220W

In Bessel (1 atm)

Detector, DAQ system,
Storage, Thermometer,
Pressure gauge,
GPS, Clinometer

Out of Bessel

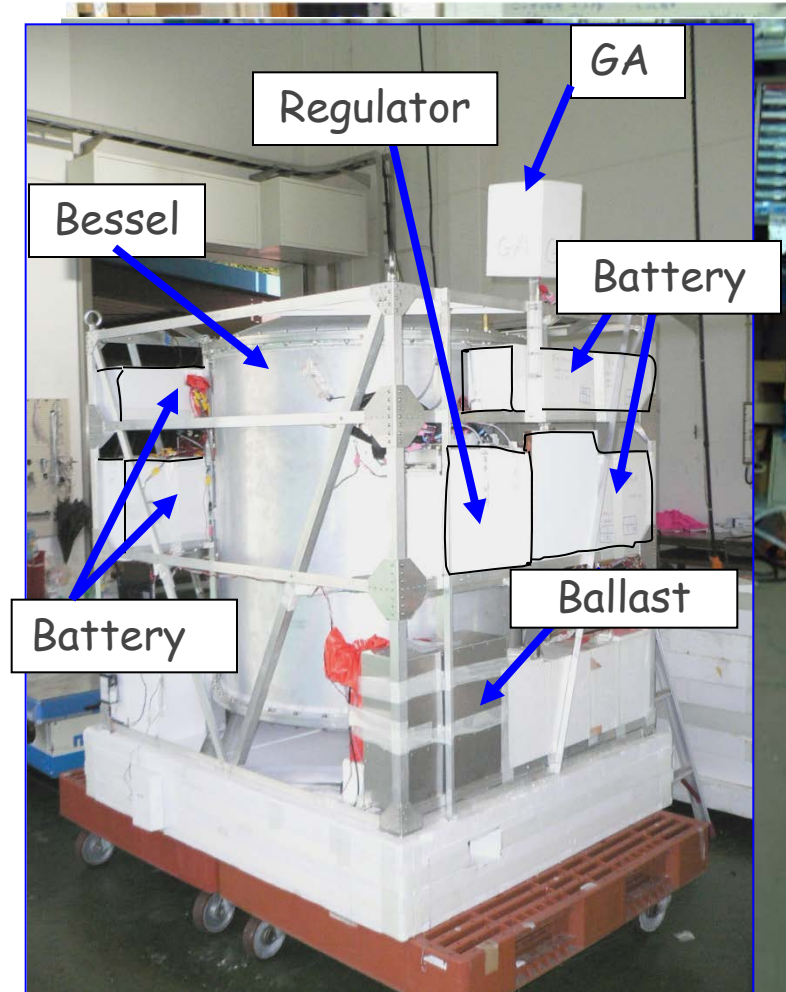
Battery & Regulator,
Thermometer,
Pressure gauge,
GPS antenna,
Geomagnetic aspectmeter

Flight Control

Telemetry,
Transponder,
Buoy, Radiosonde,
GPS, Thermometer,
Pressure gauge, etc.

Balloon

B100 ($100,000 \text{m}^3$)
Weight 816kg
Buoyancy 888.2kg



1st Flight SMILE

- Gondola size: $1.45 \times 1.2 \times 1.55 \text{m}^3$
- Gondola weight: 397kg
- Bessel: $\phi 1 \times 1.4 \text{m}^3$
- Power: $\sim 350 \text{W}$
in Bessel : 220W

In Bessel (1 atm)

Detector, DAQ system,
Storage, Thermometer,
Pressure gauge,
GPS, Clinometer

Out of Bessel

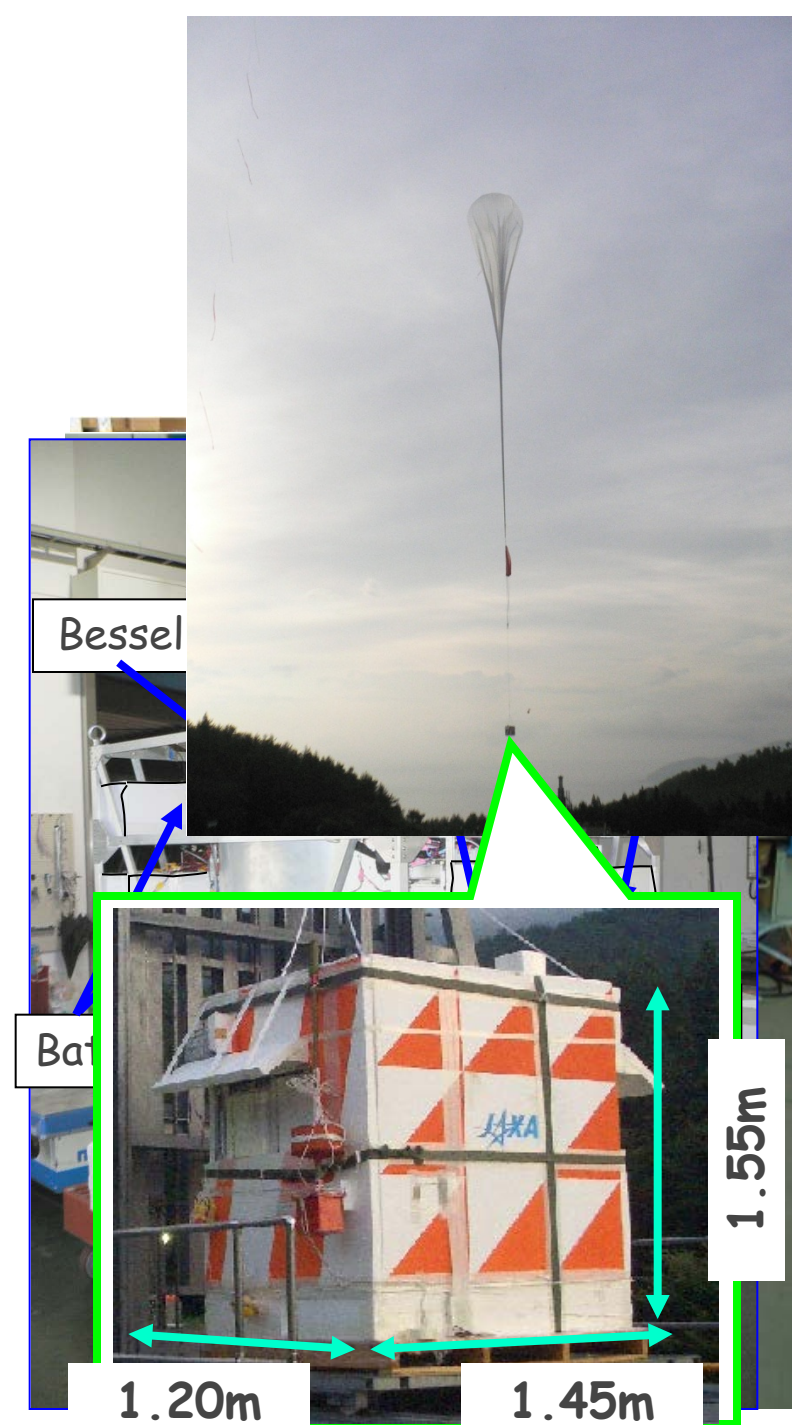
Battery & Regulator,
Thermometer,
Pressure gauge,
GPS antenna,
Geomagnetic aspectmeter

Flight Control

Telemetry,
Transponder,
Buoy, Radiosonde,
GPS, Thermometer,
Pressure gauge, etc.

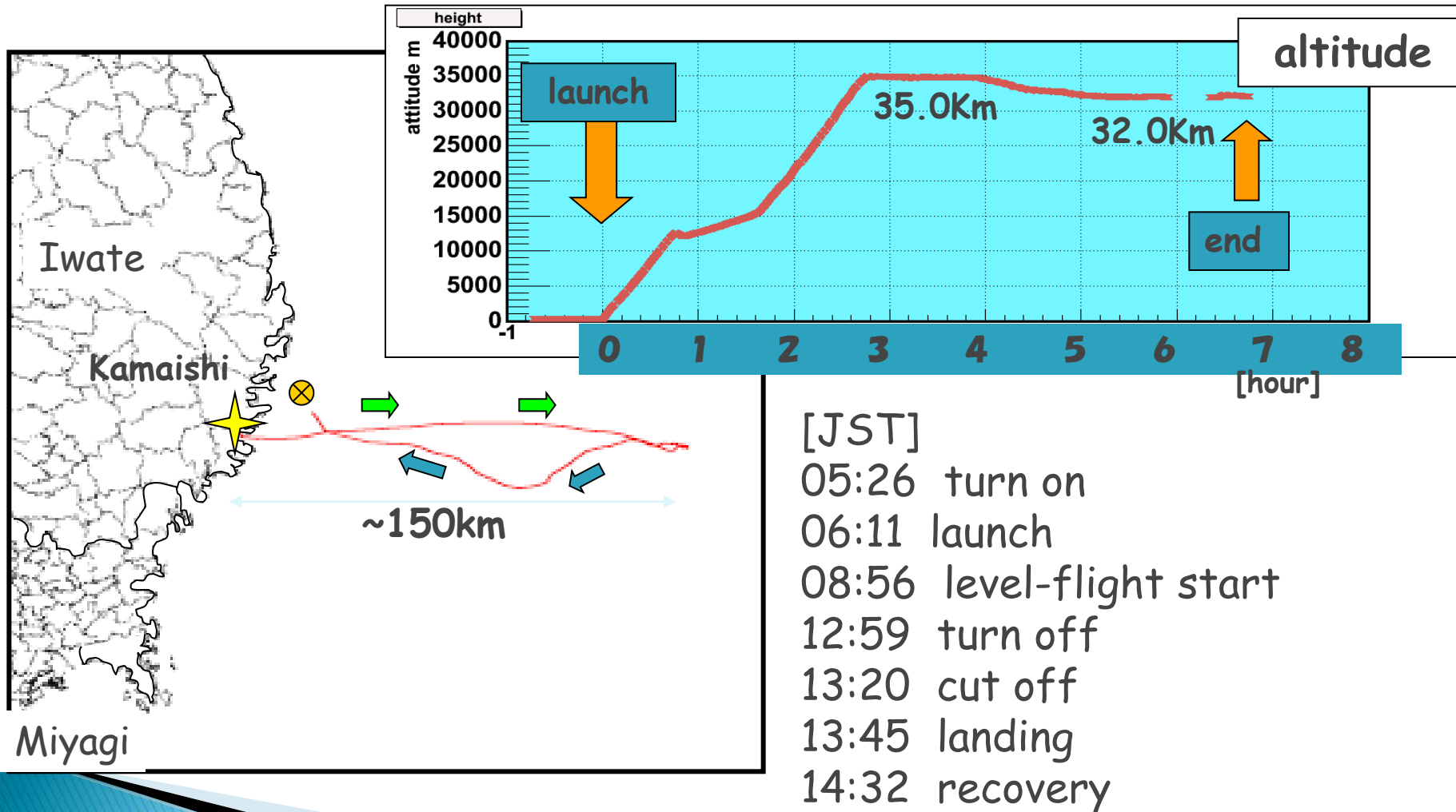
Balloon

B100 ($100,000 \text{m}^3$)
Weight 816kg
Buoyancy 888.2kg



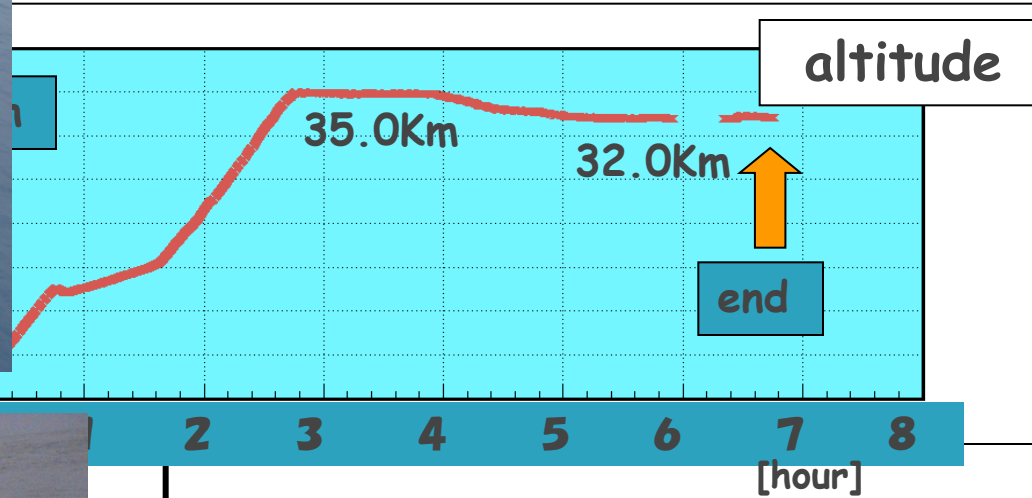
1st Flight

- ▶ Sanriku Balloon Center (JAXA)
- ▶ Launch at Sep. 1st 2006



There was no serious trouble during this flight !

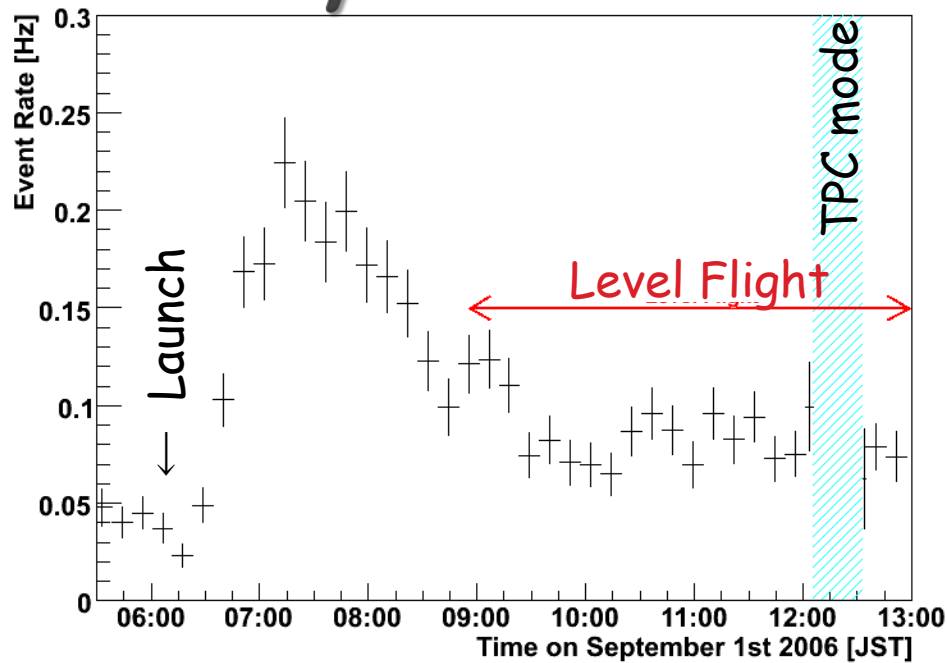
Ku Balloon Center (JAXA) Flight at Sep. 1st 2006



[JST]
05:26 turn on
06:11 launch
08:56 level-flight start
12:59 turn off
13:20 cut off
13:45 landing
14:32 recovery

was no serious trouble during this flight !

Compton event rate & spectrum

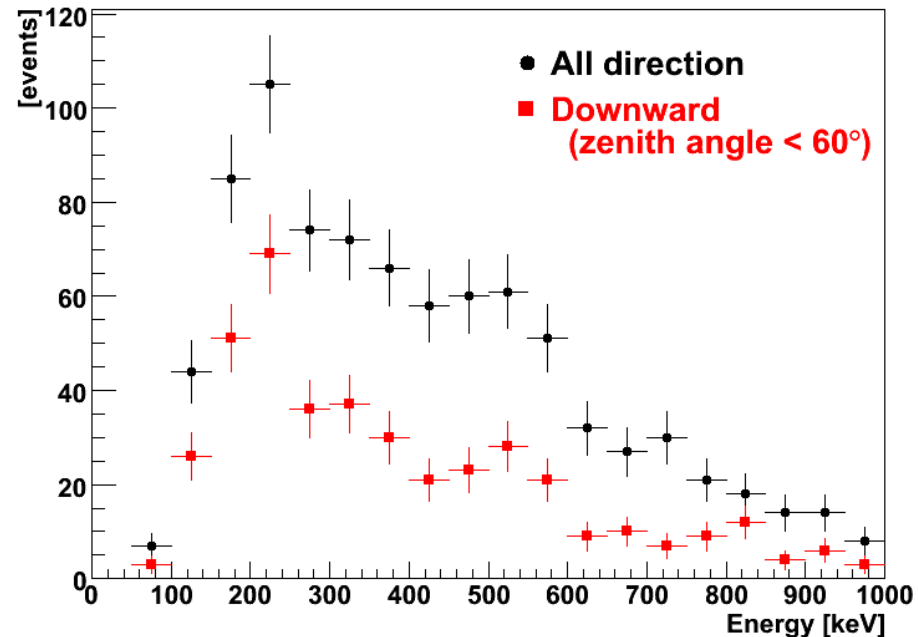


Rate of Compton event

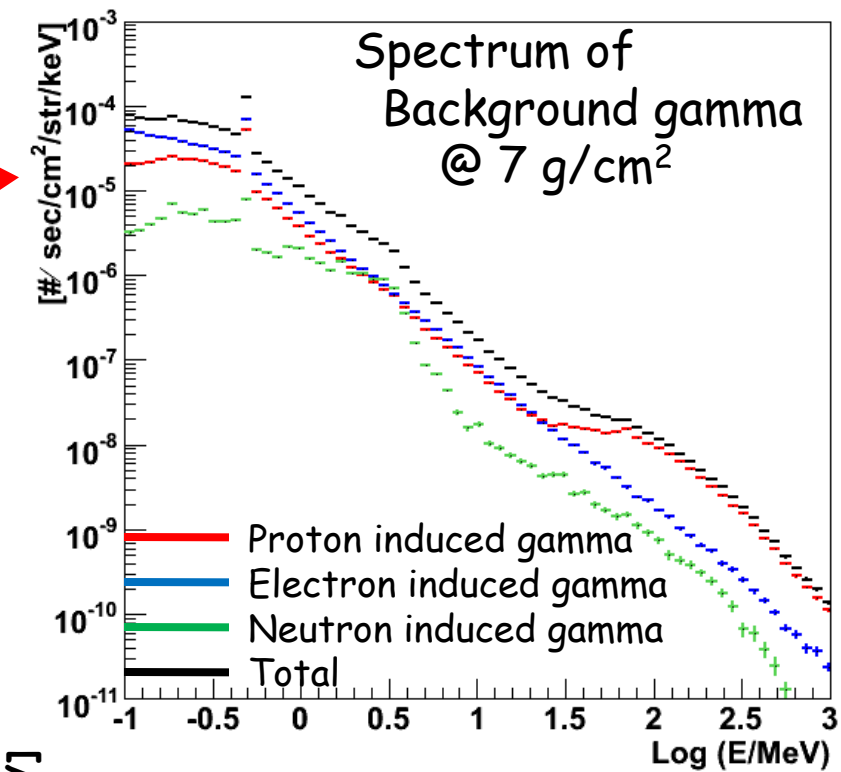
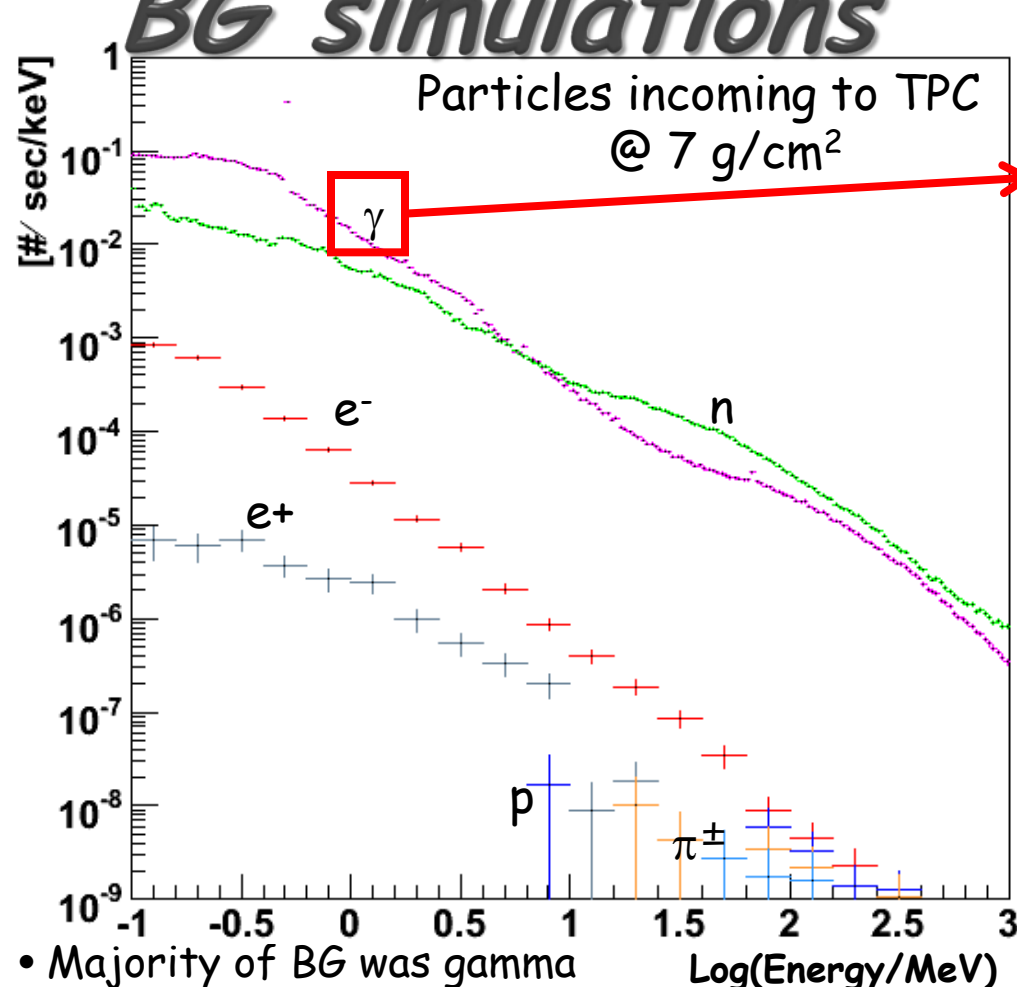
- 100~900 keV
- All direction ~2000
- in FOV (3 str) ~940

Energy Spectrum

- 32~35 km level flight
- 3.5 hours (live ~3h)
- in FOV event
~420 events
GEANT4 \Rightarrow ~400events

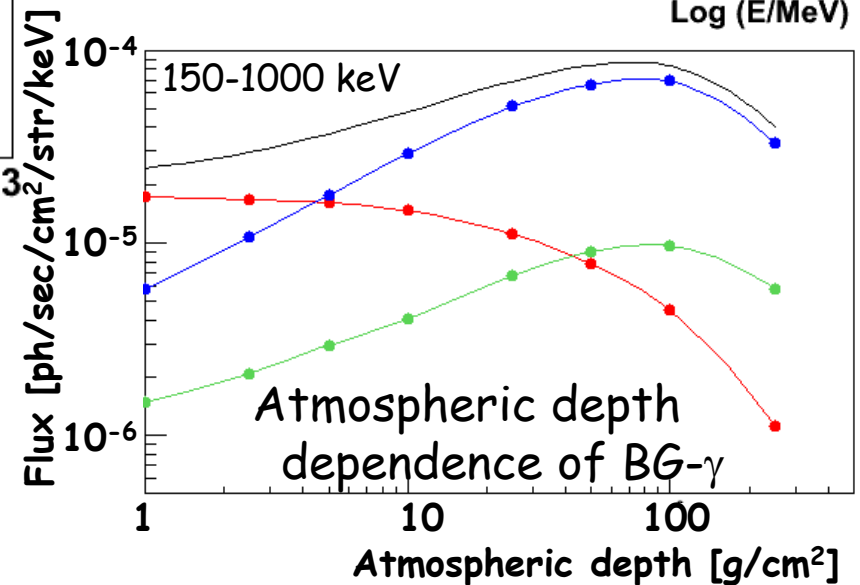


BG simulations

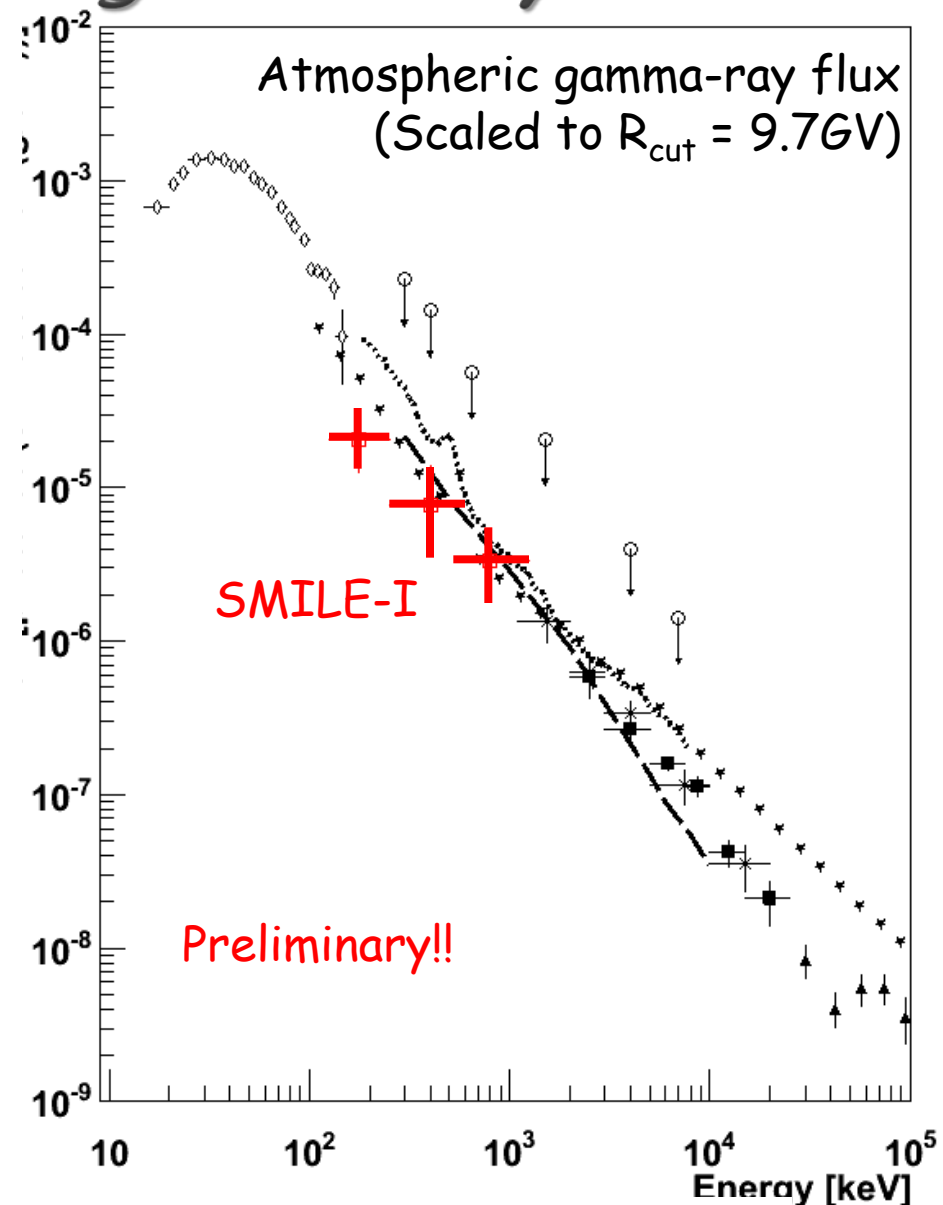
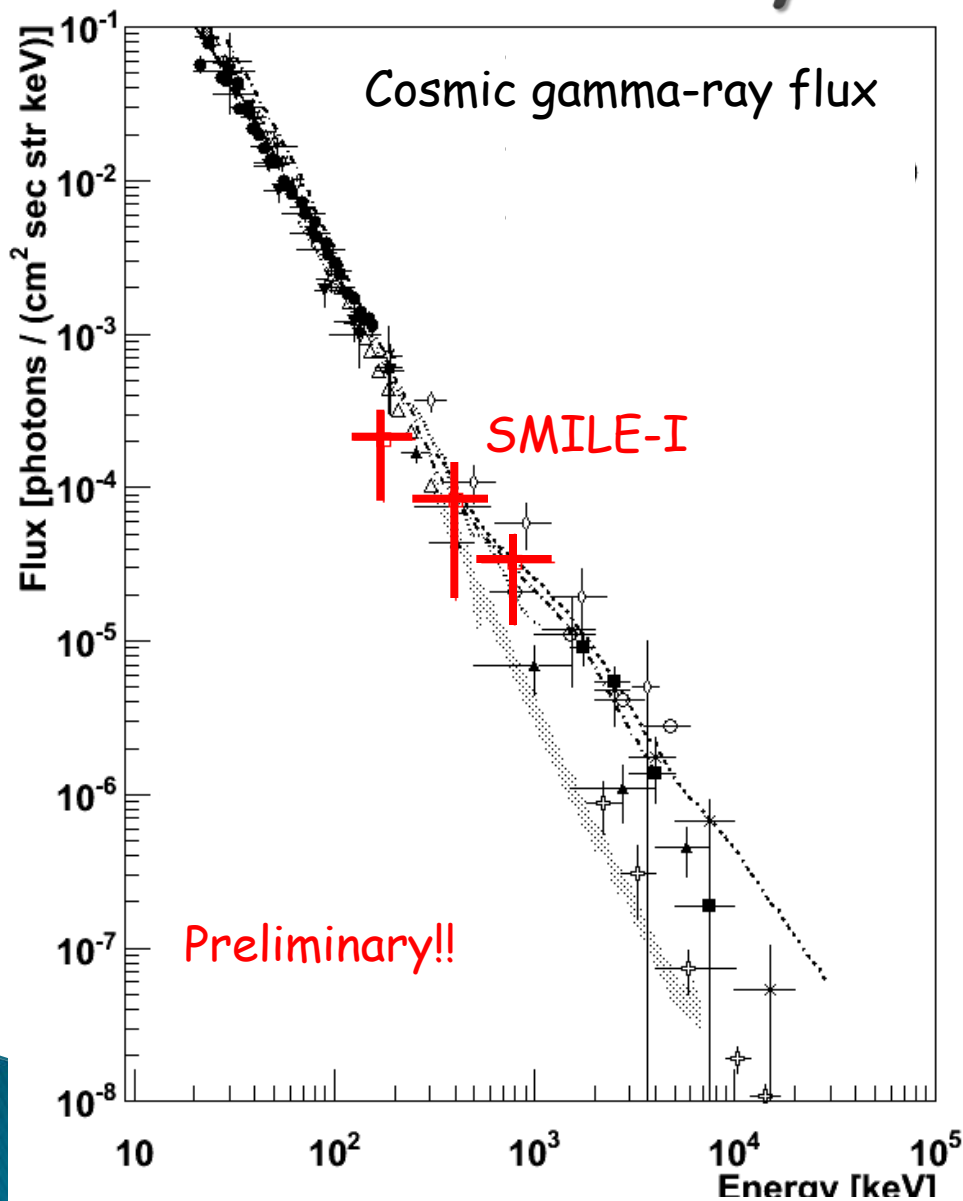


- Majority of BG was gamma produced in the gondola (BG-gamma).
- Simulator expected :
obtained Compton events at level flight

gamma-ray	~78%
BG-gamma	~20 %
neutron	1.5%
charged particle	< 0.25%

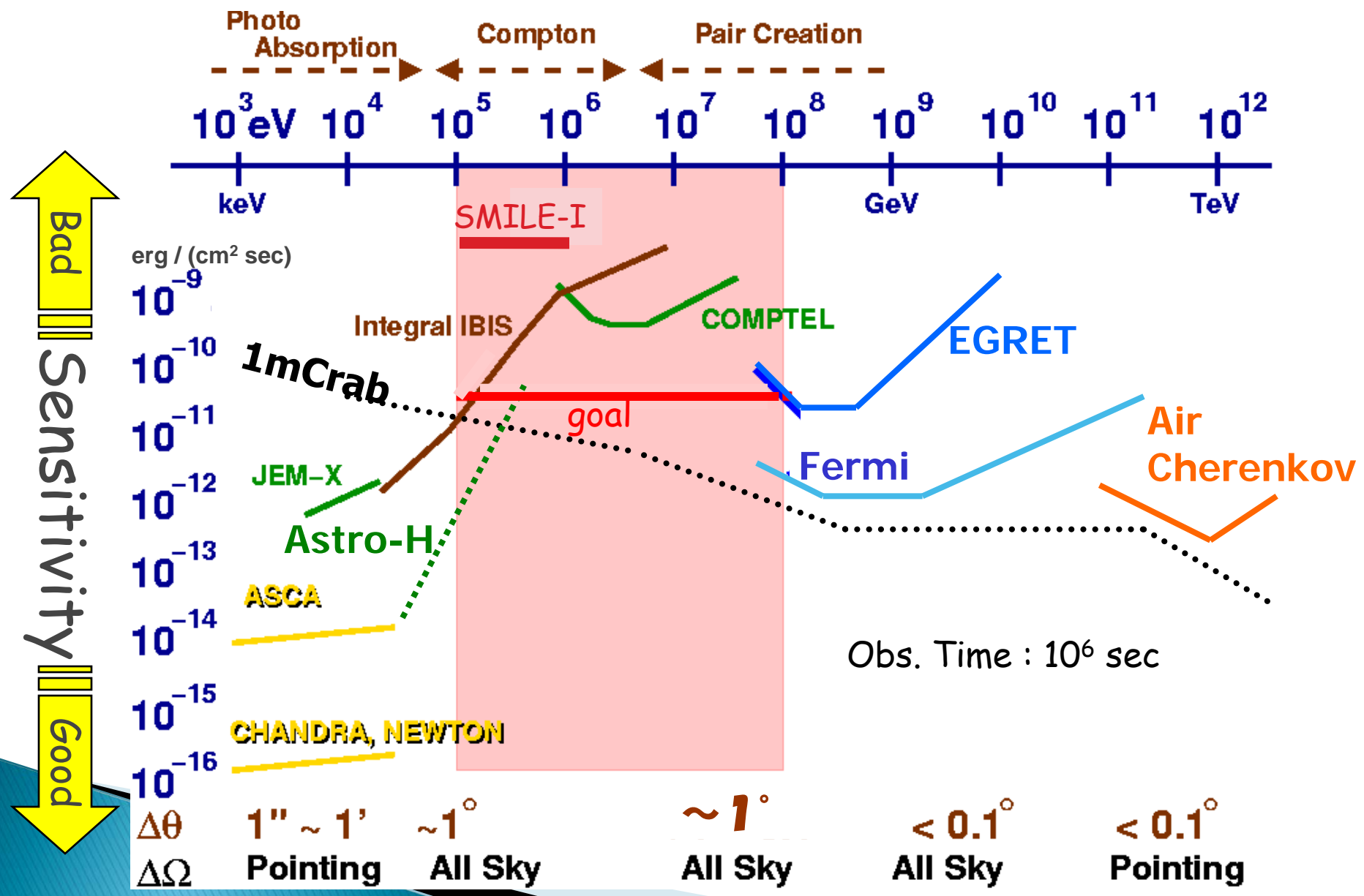


Cosmic & Atmospheric gamma-ray flux



Our results were consistent with those of past observations!!!

Sensitivity of X/Gamma-ray observations



Sub-MeV gamma-ray Imaging Loaded-on-balloon Experiment

10cm cube camera @ Sanriku (Sep. 1st 2006)

- Operation test @ balloon altitude
- Observation of diffuse cosmic/atmospheric gamma
~400 photons during 3 hours
(100 keV~1MeV)

30cm cube camera

- Observation of Crab/Cyg X-1

40cm cube camera Sub-MeV ~ MeV

- Long duration observation with super pressure balloon
- Adding pair-creation mode

50cm cube camera

- All sky survey (load on a satellite)

Toward Next Step

➤ SMILE-I : 1st Sep. 2006 launched

- Observation of diffuse cosmic/atmospheric gamma-rays
→ detection by integration in a large FOV
- Electron Tracker : $10 \times 10 \times 15 \text{ cm}^3$, Xe+Ar 1atm
- Absorber : $15 \times 15 \times 1.3 \text{ cm}^3$ @ Bottom
 $15 \times 10 \times 1.3 \text{ cm}^3 \times 4$ @ Side



Effective area : $\sim 2 \times 10^{-2} \text{ cm}^2$

➤ SMILE-II

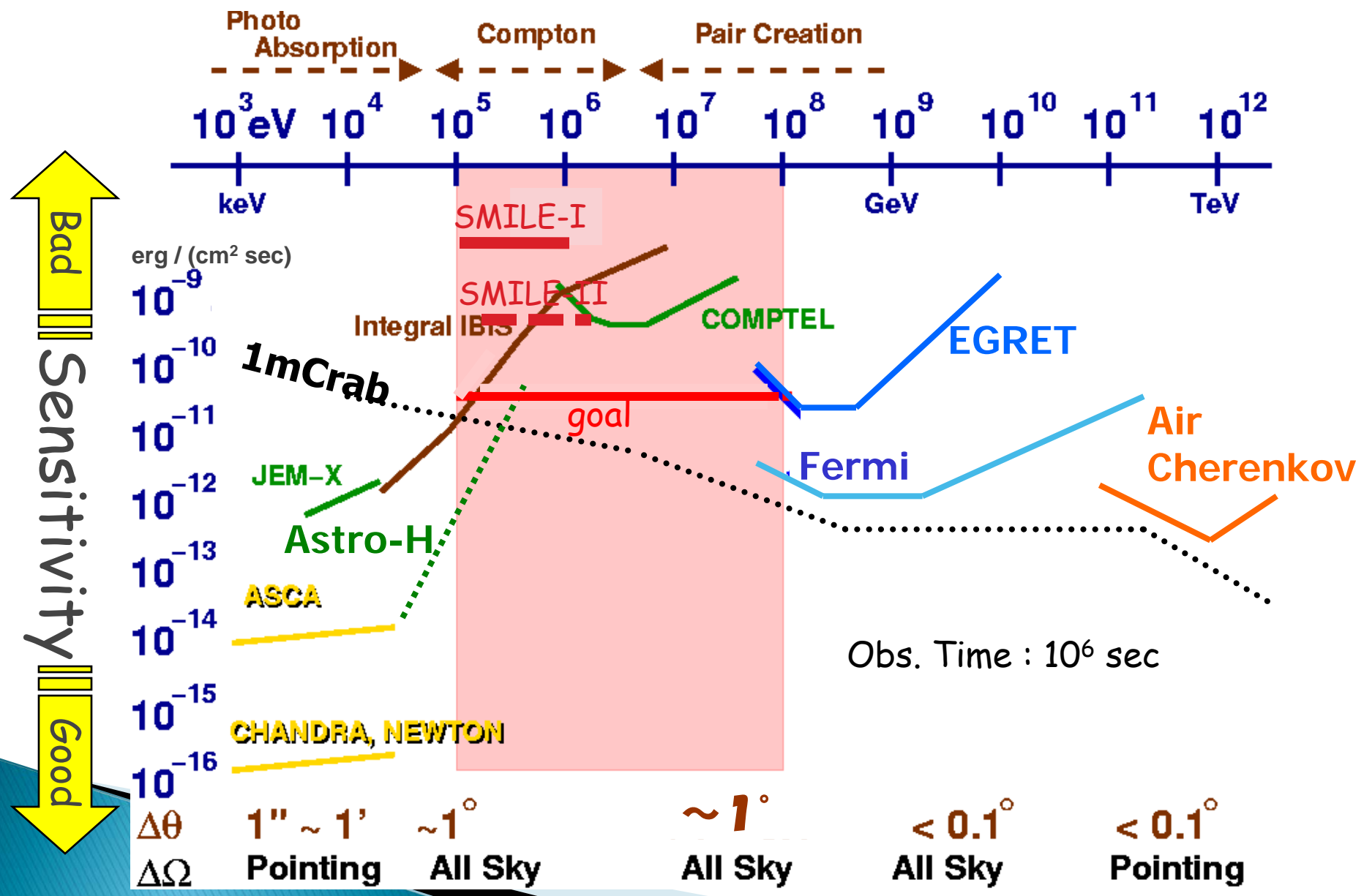
- Observation of a Bright object (Crab nebula or Cyg X-1)



Requirement : $\sim 1 \text{ cm}^2$

- Electron Tracker : $30 \times 30 \times 30 \text{ cm}^3$, Ar/CF₄ 1.5atm
- Absorber : $30 \times 30 \times 1.3 \text{ cm}^3$ @ Bottom
 $30 \times 15 \times 1.3 \text{ cm}^3 \times 4$ @ Side
- Improvement of Angular resolution

Sensitivity of X/Gamma-ray observations



γ -ray burst due to Relativistic Electron Precipitation in 1996 @Kiruna for SMILE-II

K.R.Lorentzen et al.,(2000)

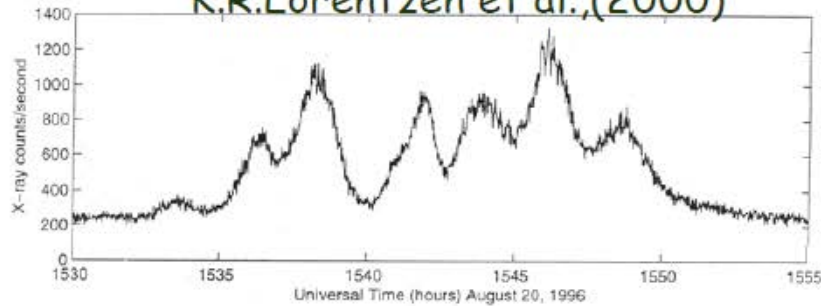


Figure 1. X-ray imager data taken during the relativistic electron precipitation event of August 20, 1996. The X-ray count rate between 20 and 120 keV is averaged over 1 s. The 10–20 s modulation is most clearly visible superposed on the peak starting near 1545 UT.

- Similar scale burst
SIMILE-II(30x30x30cm ETCC)
100keV-2MeV

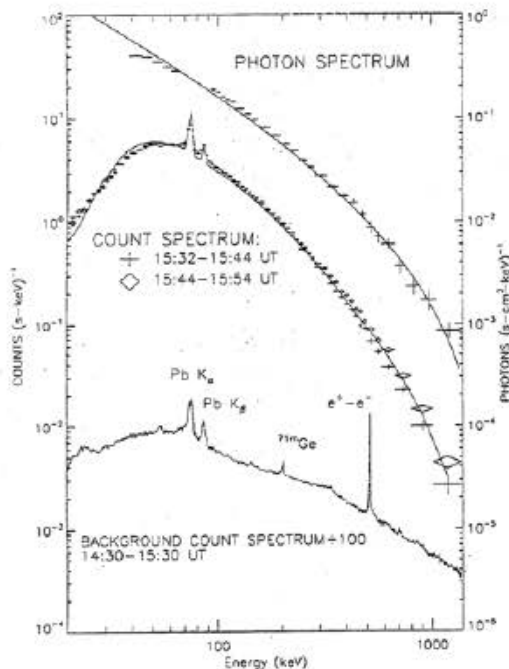
$\sim 20\sigma$ detection for imaging $\Delta\theta 10^\circ$

Good Spectroscopy from large crystal arrays.

Wide field of View with $\sim 3\text{sr}$

- Fixed point observation
 - \rightarrow spatial or temporal

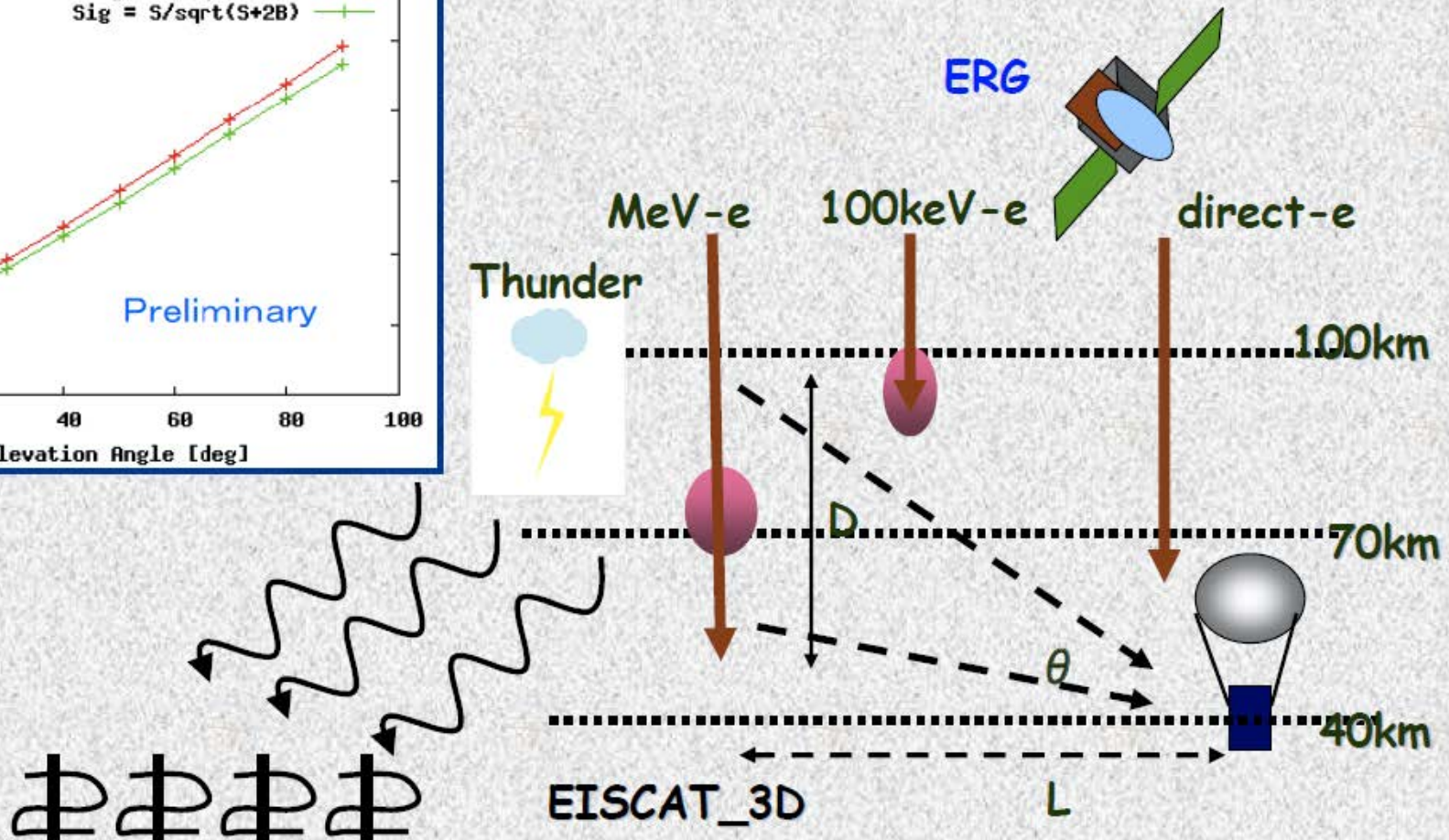
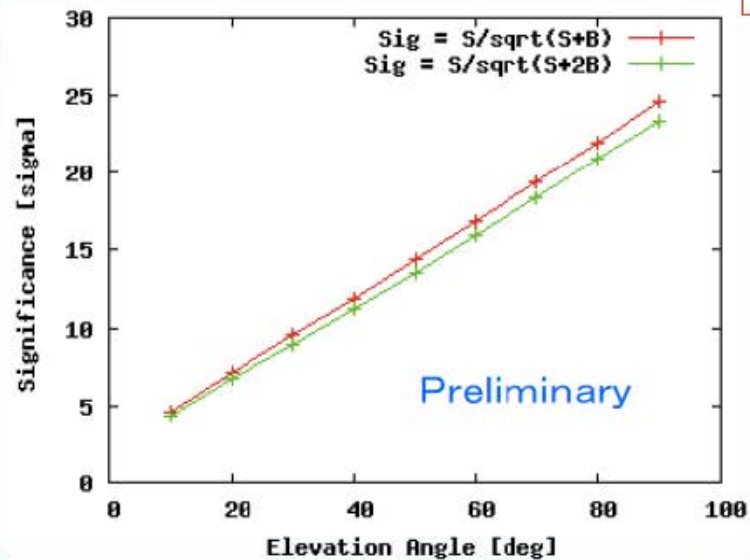
- Direct Measurements of high energy electrons, proton, neutron and nucleus



Rep-burst observation

Wide FoV imaging -> Direction, Position
Spectroscopy, Light Curve,

γ -ray spectrum -> Depth of burst
 D & θ -> L (Distance)



30×30×30cm³ ETCC current status

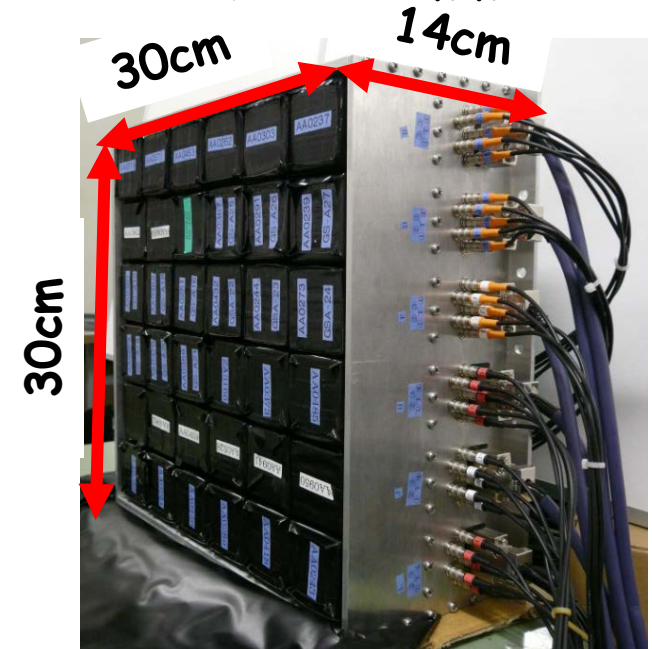
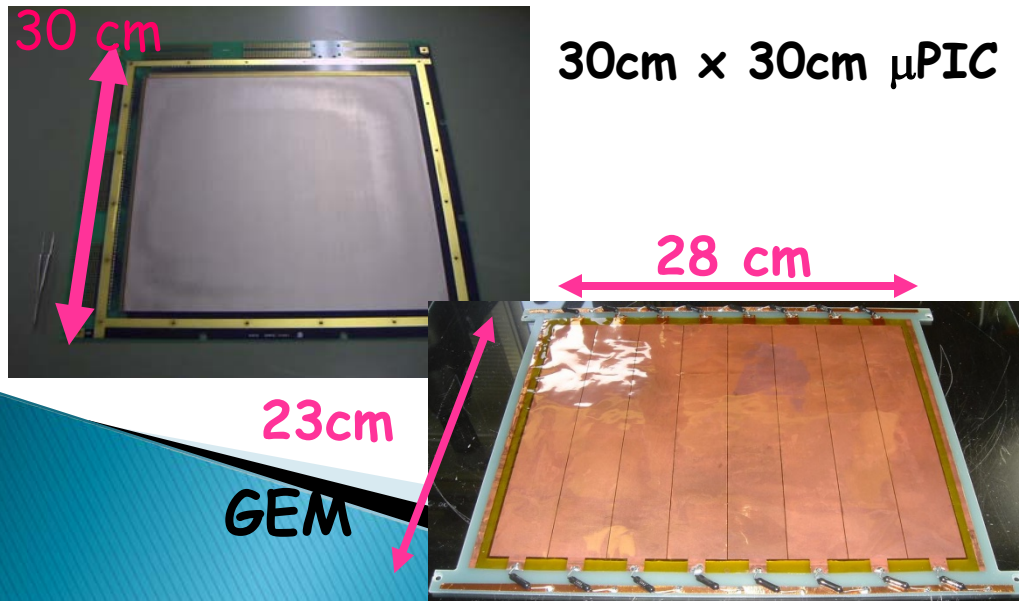
We are developing a larger ETCC based on the 30cm × 30cm × 30cm TPC and 6 × 6 scintillation cameras.

➤ Gaseous TPC

- volume : 30×30×30 cm³
- gas : Ar 90% + C₂H₆10% (1atm)
- drift velocity : 4 cm/μsec
- gain : ~30000
- energy resolution : 46%@32keV
- position resolution: 400μm

➤ Scintillation Camera

- number of pixels : 2304 pixels
- Crystal : GSO(Ce)
- pixel size : 6 × 6 × 13mm³
- energy resolution : 10.9%
(@662keV, FWHM)
- position resolution : 6mm



30×30×30cm³ ETCC current status

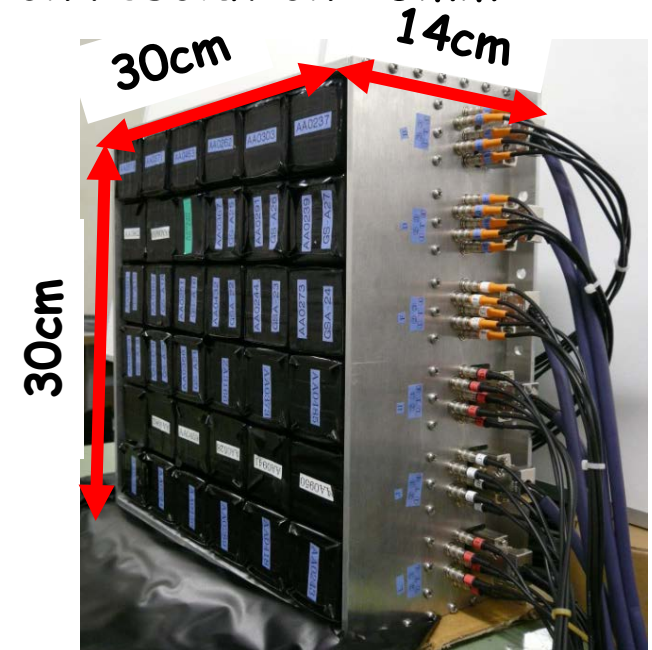
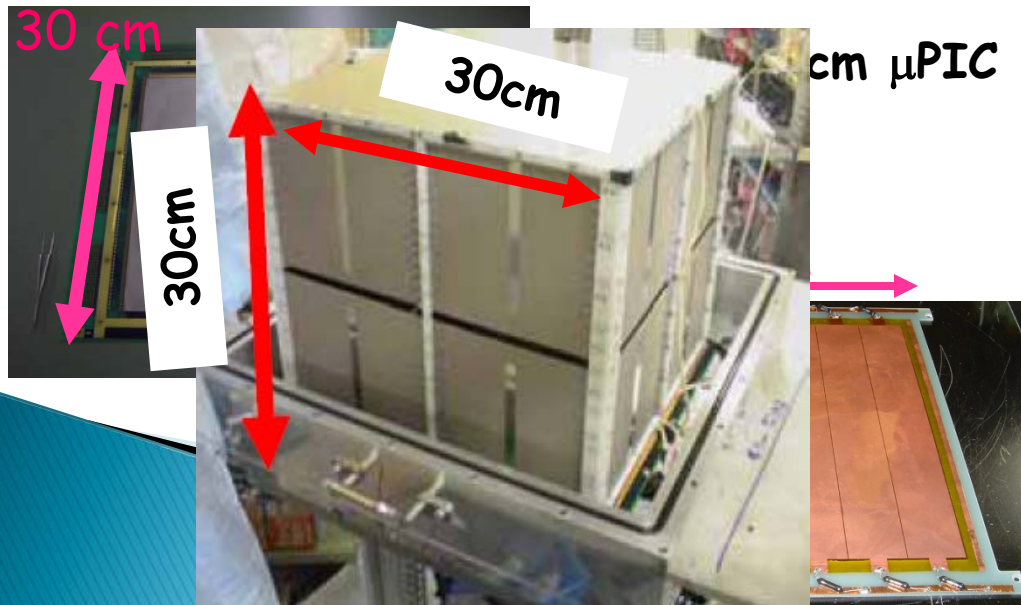
We are developing a larger ETCC based on the 30cm × 30cm × 30cm TPC and 6 × 6 scintillation cameras.

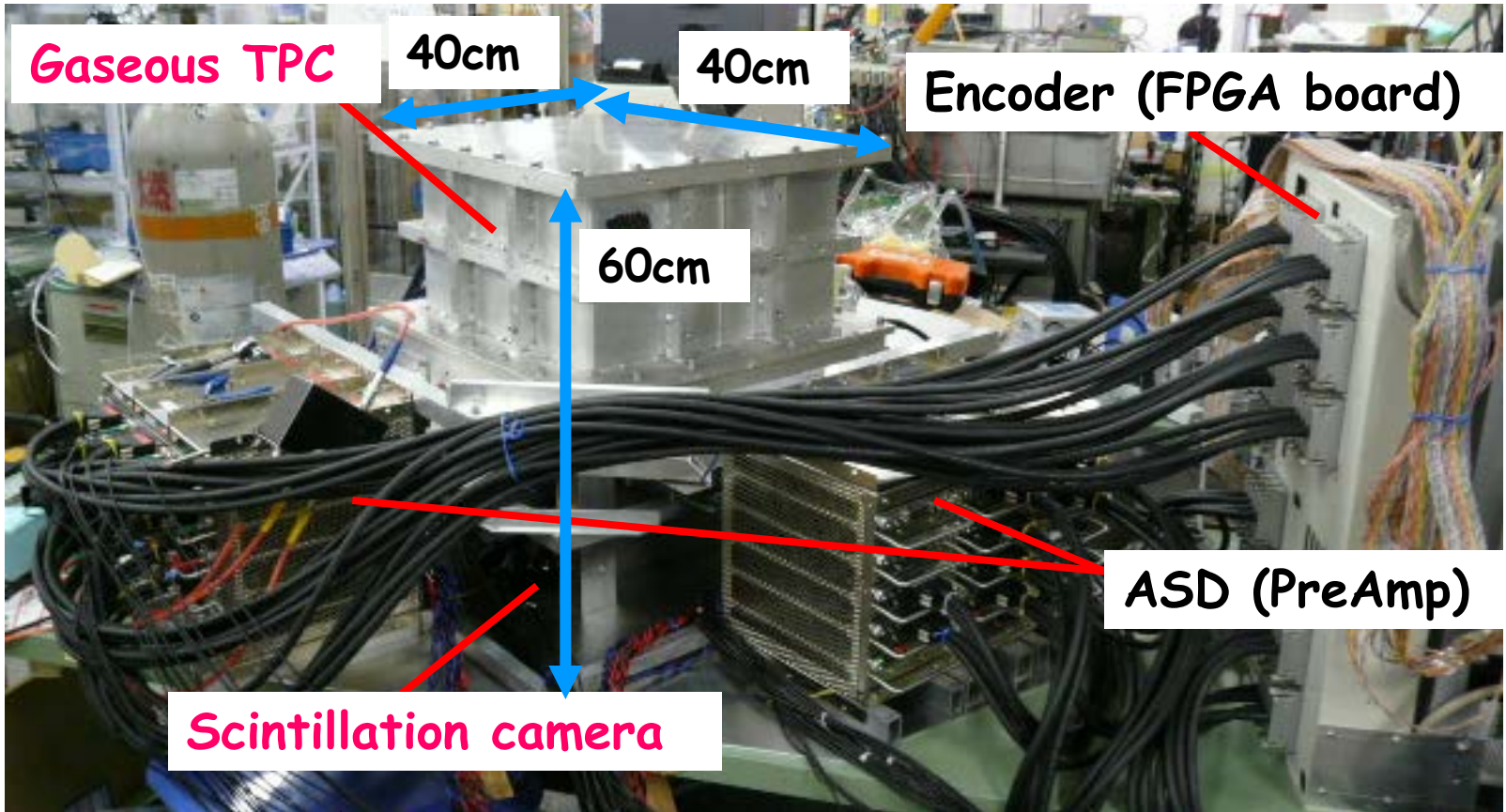
➤ Gaseous TPC

- volume : **30×30×30 cm³**
- gas : Ar 90% + C₂H₆10% (1atm)
- drift velocity : 4 cm/μsec
- gain : ~30000
- energy resolution : 46%@32keV
- position resolution: 400μm

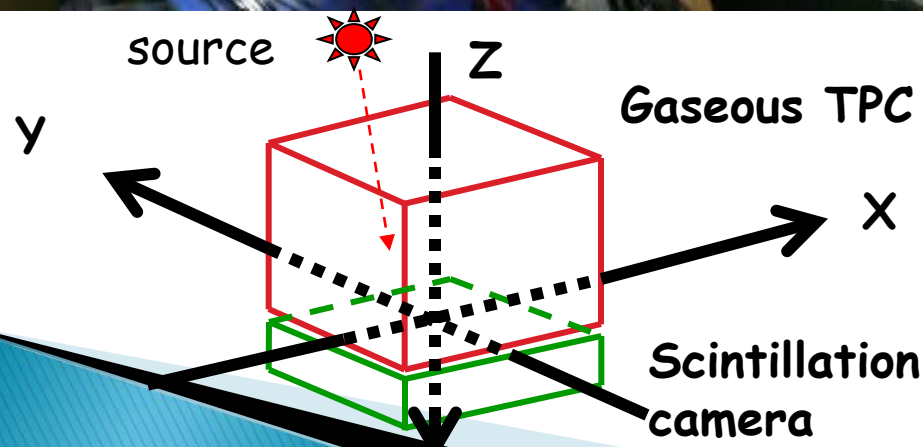
➤ Scintillation Camera

- number of pixels : 2304 pixels
- Crystal : GSO(Ce)
- pixel size : 6 × 6 × 13mm³
- energy resolution : 10.9%
(@662keV, FWHM)
- position resolution : 6mm





Setup

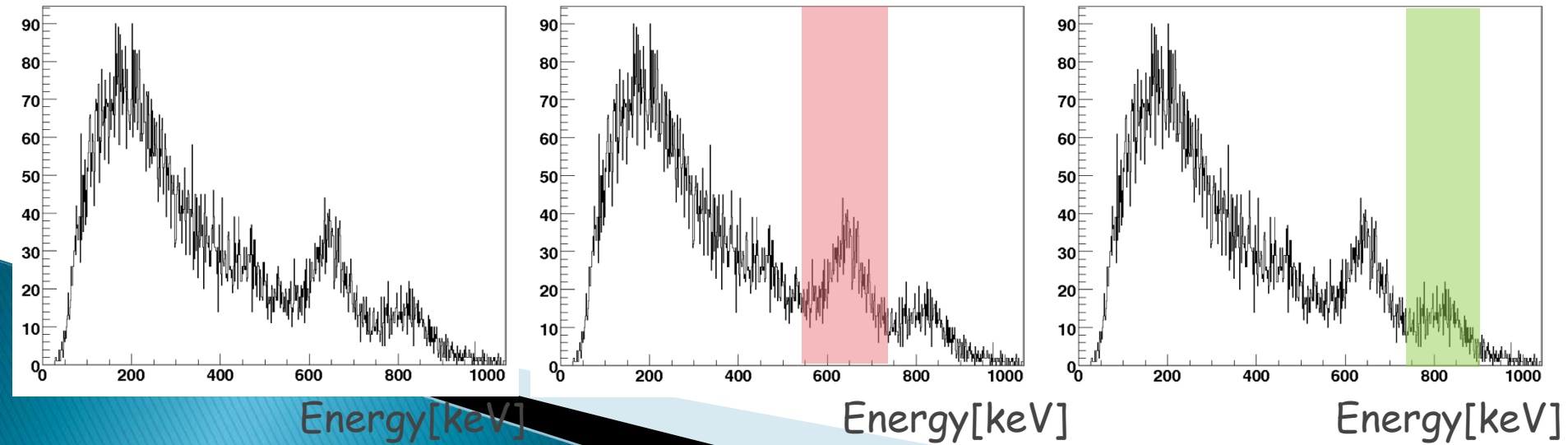
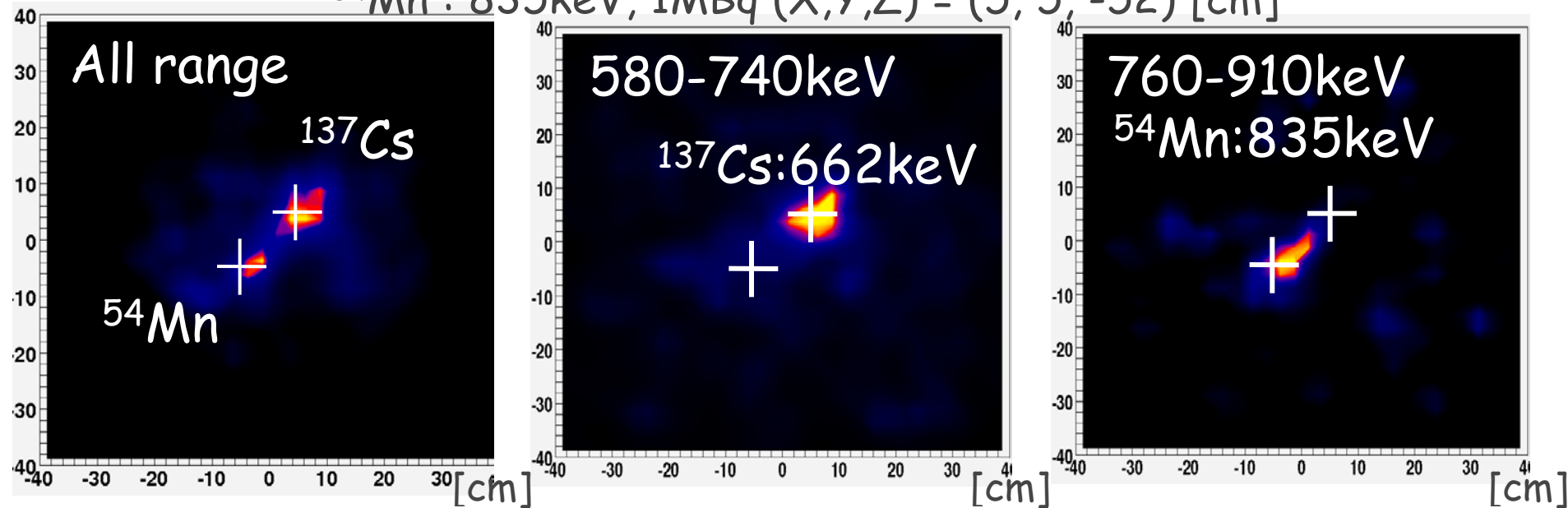


Center of μ PIC : (0,0,0)
 Center of Scinti.
 : (-3.3, 0.2, 5.7)

simultaneous imaging (preliminary)

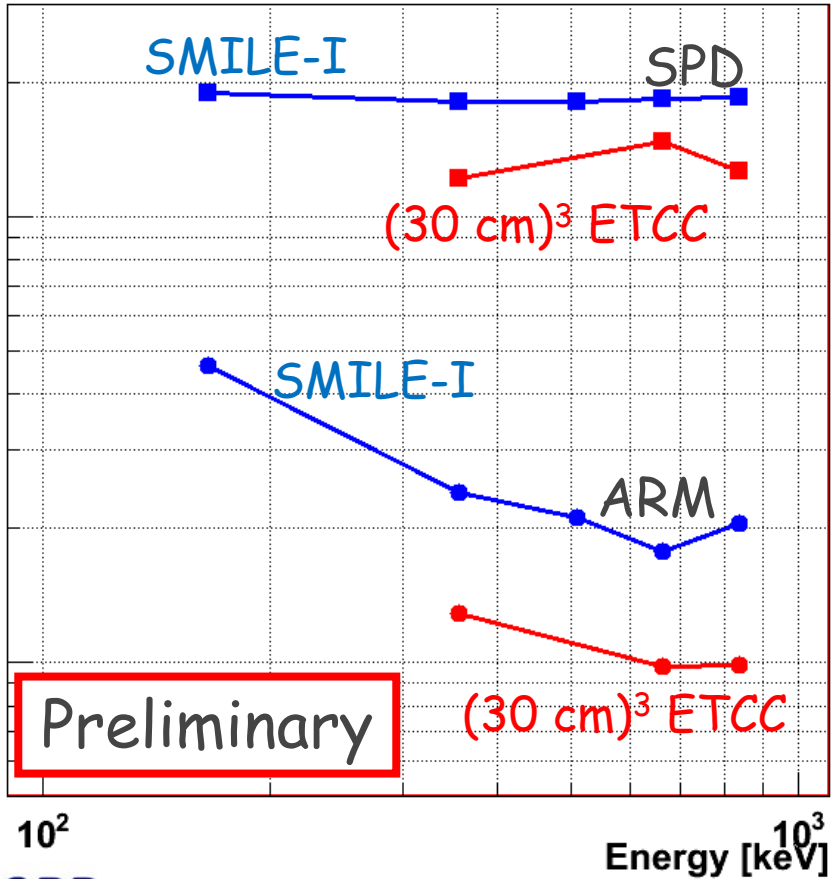
^{137}Cs : 662keV, 1MBq (X,Y,Z) = (5, -5, -52) [cm]

^{54}Mn : 835keV, 1MBq (X,Y,Z) = (5, 5, -52) [cm]

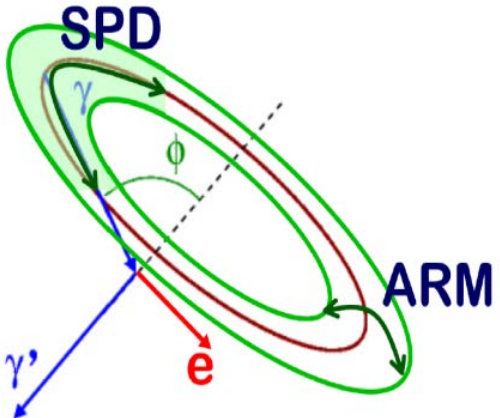
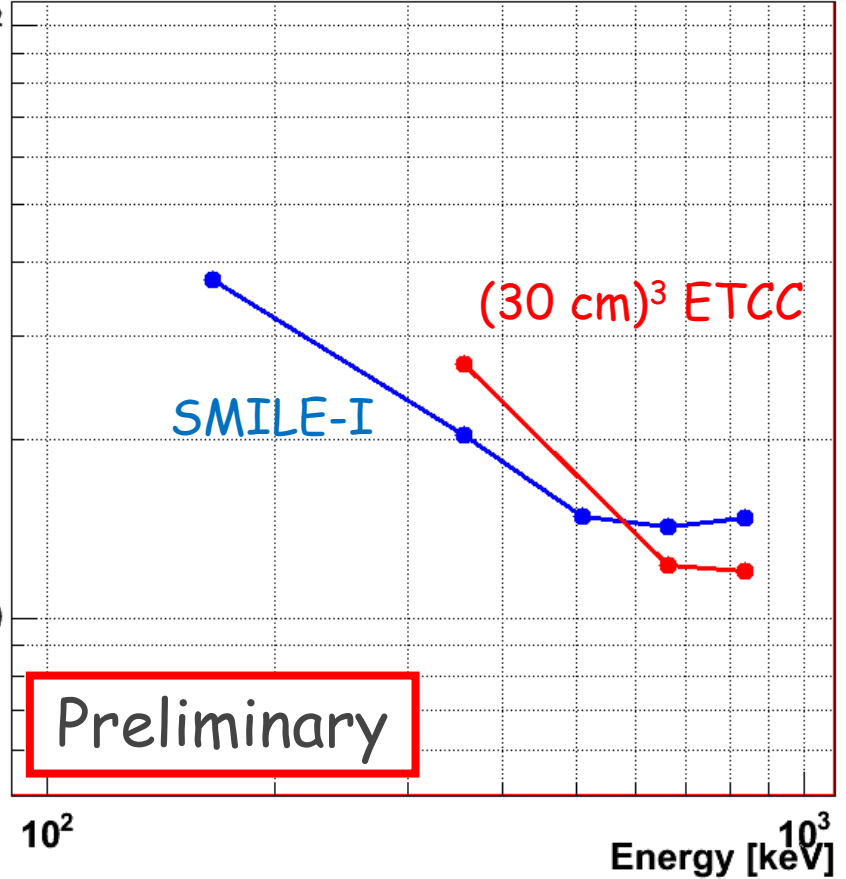


Angular resolution, Energy resolution

Angular resolution @ FWHM [degree]



Energy resolution @ FWHM [%]



SPD: 147 [deg]	↔	183 [deg]
ARM: 9.8 [deg]		17.7 [deg]
DE/E: 12.3%		14.3%
(FWHM) @662keV		SMILE-I

ARM : Angular Resolution Measure
 SPD : Scatter Plane Deviation

Saving power consumption of the readout

SMILE-I The power of readout system

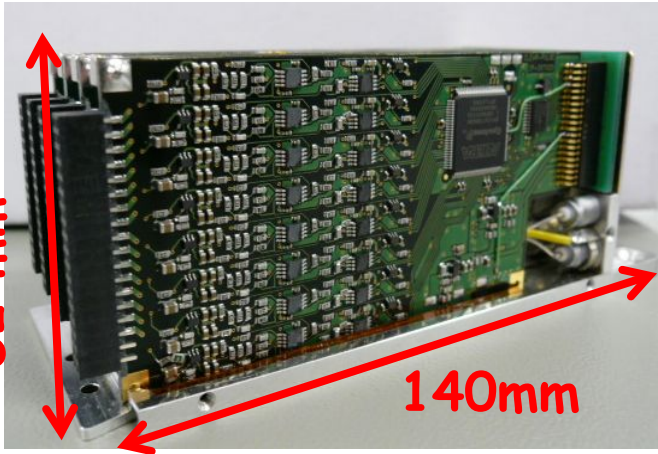
SMILE-II

33 PMTs : ~80 W
 (10 cm)³ μ -PIC (1024ch) : ~70 W

~200 PMTs
 (30 cm)³ μ -PIC (1536ch)



➤ For scintillation camera (CP80190 Clear Pulse)

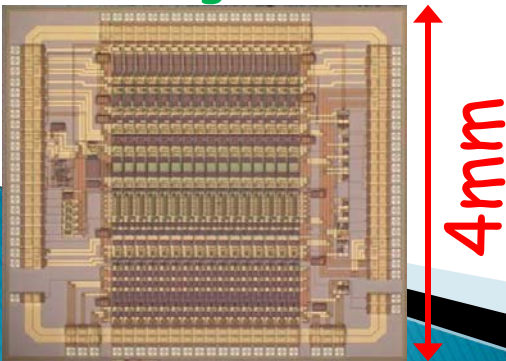


	GSO array $\Delta E / E$ (FWHM @ 662 keV)	Power (/PMT)
SMILE-I system	11 %	2700 mW
New system (SMILE-II)	10.5 %	100 mW



Collaborator: M. Tanaka, and Y. Fujita (KEK)

➤ ASIC for gaseous TPC with a 0.5 μ m-CMOS



	TPC $\Delta E / E$ (FWHM @ 22 keV)	Power (/ch)	ch # (/chip)
SMILE-I	~ 20 %	59 mW	4
New	~ 20 %	18 mW	16



Developments around detector

▶ Batteries

- solar panel and Li-ion batteries for long duration balloon

▶ Attitude control

- azimuthal control for tracking of celestial object and solar
- sensors for knowledge of gondola attitude

▶ Design of structure

- Design of gondola
 - for mechanical stability during balloon flight
 - for minimal damage of detector and sensors at the landing

▶ Thermal design

- calculation of thermal balance

▶ Telemetry and Command

- monitoring and control of all systems at anytime

Batteries

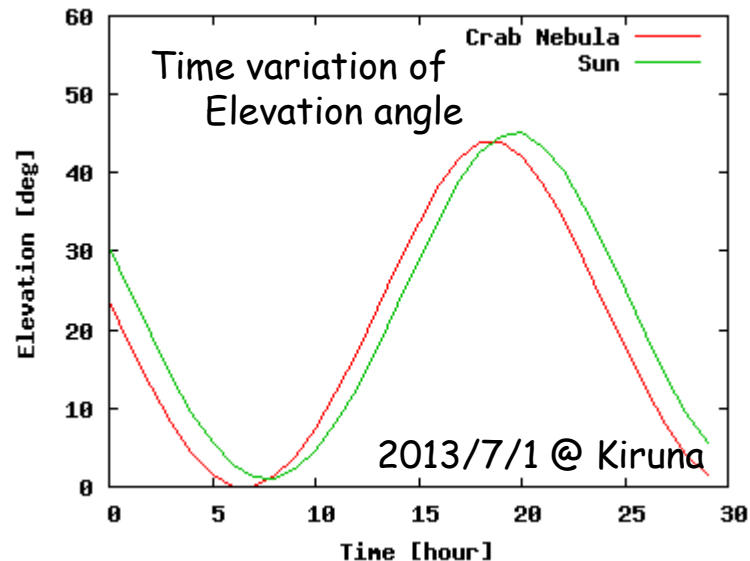
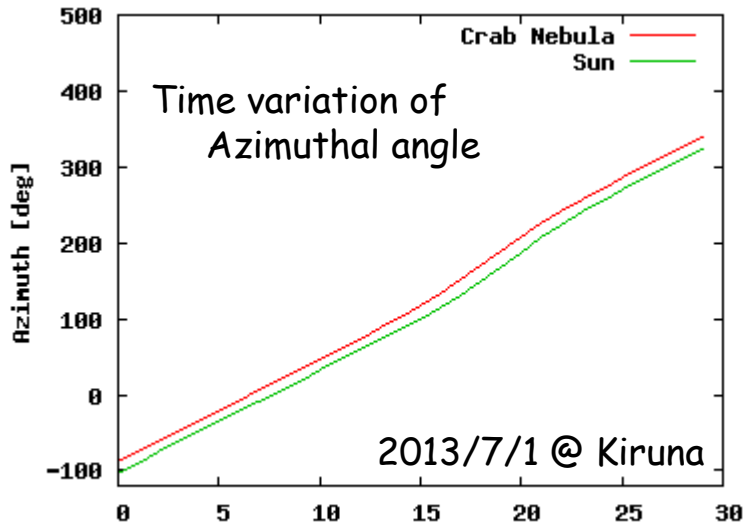
SMILE-I

- only Li batteries (primary chemistry)
- Batteries capacity: 350W, 30h (flight + dressed rehearsal)
- Weight of batteries: 40kg (10% of gondola weight)
- All system power were generated by the regulators. (loss: 100W)

SMILE-II

- Long duration balloon (~2 weeks)
 - System will require the power (batteries output) of ~400W.
 - The elevation angle of sun depends on the local time.
-
- Solar panel and Li-ion batteries system
 - Charge/Discharge control system
 - System power will be generated by the DC to DC converters.
(candidate of DC/DC: COSEL SFS series)

Attitude control



- The observation target of SMILE-II will be Crab nebula.
- The azimuthal and elevation angle of celestial objects are changing by time and position.
- Thus, we need to trace the target.

SMILE-II

Field of View : ~ 3 sr

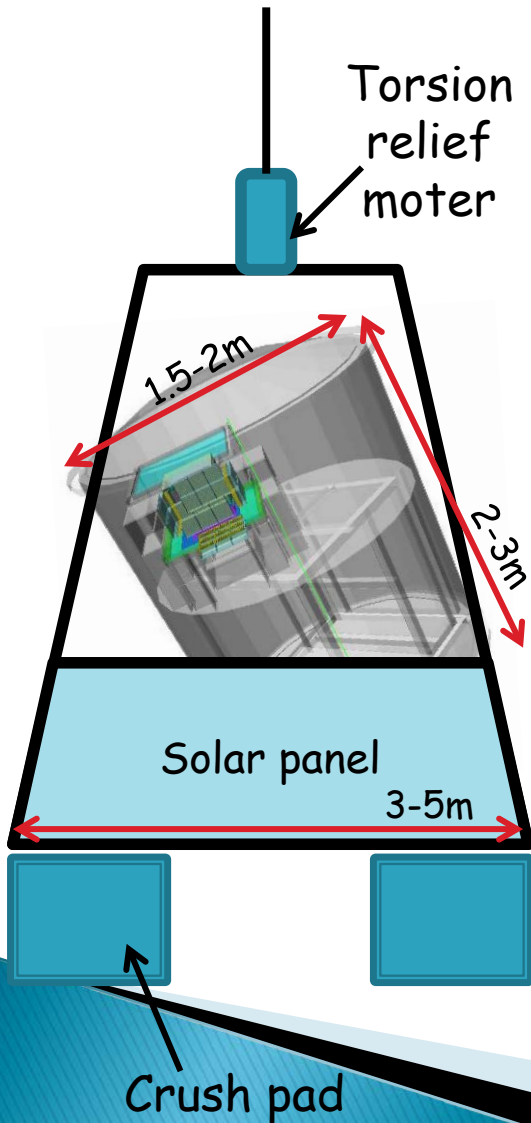
Angular resolution : $5-10^\circ$



Requirements

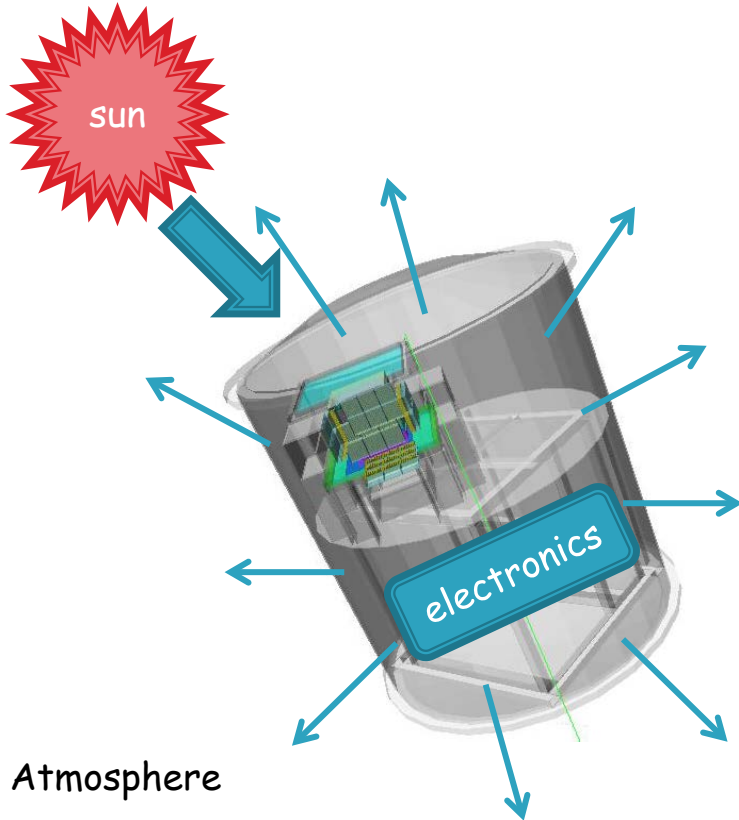
- The resolution of the attitude knowledge will be $\sim 0.1^\circ$.
GPS compass, clinometer, geomagnetic aspectmeters
- The accuracy of the attitude control is $\sim 10^\circ$.
only azimuthal control by the torsion relief motor

Design of structure



- The weight of gondola will be 500-700 kg.
 - The elevation angle of the pressured vessel will be 60-70 degrees.
 - The impact of the landing will be $>30 G$.
-
- All components must be fixed on the gondola frame rigidly.
 - At the landing, the gondola frame and the crush pad must absorb the impact, for the minimal damage to the systems.

Thermal design



Sources :

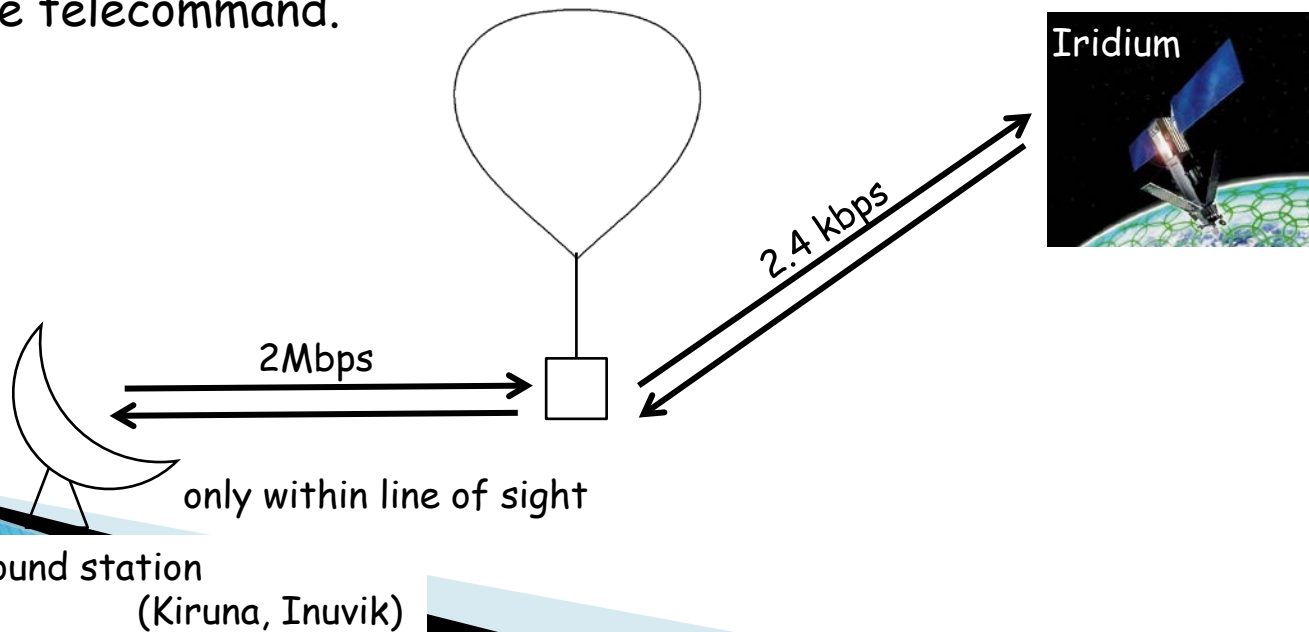
- Pressured vessel
electronics (power < 400W)
 - Sun (daytime)
 - Atmosphere
- | | | |
|-------------|--------------------|----------------------|
| | 40km | 15km |
| pressure | 3hPa | 120hPa |
| density | 4g/cm ³ | 195g/cm ³ |
| temperature | -27°C | -60°C |
- (at Taiki, Japan)

- There is a big difference of thermal input between noon and midnight.
- The electronics must be kept in the range of the operation temperature.
- The solar panel and the Li-ion batteries have a dependence on temperature.

Calculation of thermal balance is very important!

Telemetry & Command

- We must control and monitor the system at anytime.
- Our hands can not reach at the altitude of 40km, of course, we thus use telemetry and telecommand.
- Because we can not see the balloon over the horizon, we use Iridium satellite connection (2.4 kbps/channel).
- In the circumpolar balloon flight launched from Kiruna, we can use also 1Mbps downlink in Kiruna launch pad and Canada downlink station.
- We need to develop the encoder of the telemetry and decoder of the telecommand.



Thank you!

If you have any question,
please send e-mail to cr-bal@cr.scphys.kyoto-u.ac.jp

<http://www-cr.scphys.kyoto-u.ac.jp/research/MeV-gamma/en/index.html>

